

AVIATION

The Oldest American Aeronautical Magazine



"WASP JUNIOR"

*powers Major Doolittle's
Laird "Solution" in its
Transcontinental Record
Smashing Flight*



Major James H. Doolittle, who broke the transcontinental flight record

THE
PRATT & WHITNEY AIRCRAFT CO.
HARTFORD, CONNECTICUT
Division of United Aircraft & Transport
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**ECLIPSE
AVIATION
CORPORATION**

East Orange, New Jersey

(Subsidiary of General Aviation Corporation)

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October, 1931

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AVIATION

A MONTHLY PUBLICATION

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Editor: H. H. HARRIS

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Capt. R. E. S. Anderson, R.A.F., jumps out of his chute, "Life" in hand, as he parachutes from the Cleveland stadium.



Below: How Ray Ray Kaden, winner of parachute jump in Philadelphia event at National Air Races, like the climb with his chute. (at right) Major James H. Doolittle lands his "Life" in rescue of Seattle factory flame, following at Cleveland, on his successful rescue jump.

There Was Safety to LIFE --- at the Races

Safety at all times -- and, in emergencies, 4 lives saved.

In the Air Derby from coast to coast, as well as in the National Air Races themselves at Cleveland, increased confidence flew with the increased speed and performance that man and machine pilots attained. This confidence was based not so much on the fact that IRVIN Air Chutes was 9 out of 10 times as well as 6 seconds and 1 death in the Parachute Exhibition events as it was that in actual service use these "Life Preservers of the Air" had actually saved over 500 lives -- and there at the Races, added 4 more to the grand total of the Cessna Club.



IRVING AIR CHUTE CO., Inc.
312 Pearl St., Buffalo, N.Y.

West Coast Factory and Office: 1800 New Street, Shoshone, Calif. Coasting Factory: Bridgeport, Conn.

IRVIN AIR CHUTES

"The Life-Preserver of the Air"

"Heavy Landing?" our machine proves an absolute safety. Following IRVIN Air Chutes in action, no doubtless from all emergency, safety and rescue situations as shown here. (See for booklet and price, etc.)

Wholly changing public opinion of AIR TRAVEL



A TRUE REALIZATION of the possibilities of the Autogiro in the general public and the aviation industry can come only from actual knowledge of Autogiro flight.

Here is part of what Ben Ray Kaden, winner in a review of the book "Wings of Tomorrow" in the New York Times, June 18th, 1931:

"One morning last winter, standing on the Piccadilly sidewalk near Philadelphia, I watched James Ray take a flying machine into the air and mind it to drop that I knew an airplane could do. As he took off slowly, rose up and fell down, I knew he was going to crash, when he drifted over our heads at a speed of about twenty-five miles an hour, I felt a little sick, for a mill at that height was bound to be fatal, while he swayed around the curve, I looked away to avoid seeing him slip over a sign, and when he began to sail almost vertically towards the earth, I could only hope that an accident would be on the spot when he struck. But he neither stalled, spun nor crashed, and, as he settled lightly as a gull, with almost no forward speed at all, I found myself relaxing the old posture there isn't so much as usual.

"I was wrong, of course. There is, and is not, strictly speaking, an airplane. It is the Autogiro, invented by Juan de la Cerna, and it is the nearest thing to a perfect flying machine that has yet been built by man. . . . Those long rotor-blades that you may have seen whirling in the sky, and that suggest a helicopter in their motion, are really driven by no power except that of the air itself, but they turn, pushed by air currents and changing their own aerodynamic forms, they have a lifting surface that replaces the rigid wings of the ordinary airplane. They are, indeed, the "wings" of the Autogiro, for it is on them that that new machine flies, but they are free, flexible wings, quick to adapt themselves to the requirements of different speeds and different positions; and it is this new adaptability that makes the Autogiro safe, sound, and almost foolproof. . . .

This expression of enthusiasm typifies the wholly changed attitude toward air travel that results from experience with the Autogiro.

The Autogiro Company of America is not a mere manufacturing or selling company. It is really an engineering and licensing organization. It owns and controls, exclusively, all Autogiro patent rights in the United States. Manufacturing companies of high standing will be licensed to build Autogiros with the full cooperation of our engineering staff.

Present business: 1010 Airport Company, Detroit, Mich. — Kellen Aircraft Co., Philadelphia, Pa. — Patten Aircraft, Inc., Willow Grove, Pa.

AUTOGIRO COMPANY OF AMERICA • • • SAND HILL BUILDING • • • PHILADELPHIA



The Autogiro differs basically from all other heavier-than-air craft in that it can take off at low speed and a very short run, and so naturally it is a very safe machine to fly. It can fly with over 100 miles per hour at its normal 140 miles per hour. It can be brought momentarily to a standstill and then it can bank and turn slowly without fear of loss of forward speed. It can glide or descend vertically at a speed less than that of a man descending at a parachute, and with virtually no forward speed even with a dead engine. Above all, it cannot fall off one's back in a stall. As a result, full emergency, and is required.

The Autogiro presents flying characteristics that are absolutely safe. It can take off at low speed and a very short run, and so naturally it is a very safe machine to fly. It can fly with over 100 miles per hour at its normal 140 miles per hour. It can be brought momentarily to a standstill and then it can bank and turn slowly without fear of loss of forward speed. It can glide or descend vertically at a speed less than that of a man descending at a parachute, and with virtually no forward speed even with a dead engine. Above all, it cannot fall off one's back in a stall. As a result, full emergency, and is required.

SPEED+STRENGTH

of SUMMERILL TUBING

Fast planes and daring stunts at the 1931 National Air Races marked the rapid advancement in aircraft engineering. Planes entered in the different races broke all previous speed records. During stunts that require the most accurate flying, dependable engines and sturdy ships thrilled thousands of spectators.

But back of all this speed and acrobatics—making these performances possible—was the tremendous strength of Steel tubing. Each piece of tubing in these planes must be of uniform strength—welded together into a structure that will easily withstand the terrific strain of such performances. That is why Summerill Seamless Steel



Tubing is used in most of the planes you saw at the Air Races.

Summerill Tubes are made to rigid specifications and are uniform in strength. Inspection is under careful control. Back of all this is our thirty years of experience as tubing manufacturers—thirty years of making the finest quality tubing ranging from a delicate

hypodermic needle to the minute accuracy of a rifle barrel.

And so with confidence we look forward to even greater aircraft performances, because we know Summerill Tubing will be the safe, dependable backbone on which these powerful engines will fly.

SUMMERILL TUBING COMPANY

BRIDGEPORT (PHILA. DIST.), PA.

THE STRENGTH OF THE PLANE IS SUMMERILL TUBING

TUBING by SUMMERILL



ON THE AIRWAYS TODAY
as on the highways!
for the last 30 years!



"Guaranteed Forgings"

WYMAN-GORDON

Worcester, Mass. and Harvey, Ill.

FAMOUS FLIGHTS WITH THOMPSON VALVES



*This advertisement is one of a series
showing various airplane flights in
which Thompson Valves were used.*

BREAKING THE WORLD'S NON-STOP DISTANCE RECORD

(New York to Constantinople)

CARRYING 130 gallons of gasoline, 25 gallons of oil, and two fuel pilots, the Helios, with monoplane "Cape Cod" fought her way into the air against 7500 pounds of weight at Floyd Bennett Airport, New York, on July 28th last. Forty-one hours and eighteen minutes elapsed. The continuous gasoline sup-

ply was consumed to the last drop. But by that time the success of Turkey's ancient capital lay beneath the ship, and a safe landing was made at Yendiye Airbase, Istanbul—4,990 miles away.

This record-breaking flight—among the greatest in its appeal to the imagination and the thorough-

ness of its accomplishments—was made possible by the dependable performance of the "Cape Cod's" 360-hp. Wright Whirlwind motor. A vital factor in this performance was the valves—which were furnished by Thompson Products, Incorporated. THOMPSON PRODUCTS, INC., Greenfield, Cleveland Ohio, U. S. A. Phone: CLEVELAND 10-1741



AVIATION

A MONTHLY PUBLICATION OCT. 1931

The Oldest American Aeronautical Magazine

EDWARD F. WARNER, Editor

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Looking toward the Budget

FOR any government, as for an individual, sound business, a budget plan is a necessity. Only the United States among great civilized nations managed for many years to get along without one. Only our great and rapidly growing nation could make it possible for us to pursue such an inherently extravagant course, with complete lack of economic control over new projects for spending money. In 1901 we turned our back upon the old ways, and since then the Bureau of the Budget, scarcely the assurance of the President's personal financial will but probably more or less an autonomous body under the guidance of its Director, has played a great part in Washington as some budgeting authority does in every foreign capital.

The Bureau of the Budget works in silence, almost in secrecy, and it is easy to underestimate its work. Its power is awesome. It is the almost invisible power for Congress to take the budget estimates as basic-making frequent minor cuts and a few very small increases, but hardly touching the appropriation bills which differ only in detail from the budget submitted is a bulky volume accompanying the President's message on the day following the opening of the session. If the aircraft industry recognizes that there is likely to be,

on almost every statement on financial topics now coming forth from Washington would suggest, a substantial cut in the statement for the purchase of new airplanes and a substantial addition to unemployment in the aircraft industry in the name of government economy, the time for it to make its voice heard is now. We should not delay until the news of the Bureau's definite decision, as ratified by the President, is set up on the bulletin boards.

Most citizens have only the vaguest idea of what the Budget does or how it does it, but even though it be shrouded with the airs of mystery it is a human organization, subject to human limitations. It must depend on help, not its work and reaching its conclusions, on the same processes as any other individual or organization would employ. To wit, it must get its information on what is going on and what is likely to happen in future by inquiring at the source.

In most cases inquiring at the source means inquiring from the government department most directly concerned. When the Director of the Budget and his assistants seek to find out how much money they should allot for carrying on the foreign service of the United States they turn to the State Department, as the only available authority upon the magnitude of the work in hand and its probable cost. The same is true of appropriations for the support of agriculture, or the Coast Guard, or the Post Office. Inquiry from government officials is exclusively sufficient, and the budgetary

authorities ordinarily confine themselves to order inquiry, but in considering the appropriations to be made for aeronautical activities, especially for the purchase of aircraft, the military and naval use, the interested government departments alone are not in the position to tell the full story.

There is another vital interest outside their immediate scope—that of the aircraft industry. The industry's development and status have long been recognized as matters of national concern. They constitute an essential part of the material to be considered by the Director of the Budget in preparing his statement for the coming fiscal year for the consideration of the President and transmission to Congress. The government cannot afford to let the industry decline too low. Its position, and its opinions, cannot be completely or properly estimated by military and naval personnel alone. (The Budget Bureau's position of holding its meetings behind closed doors, with witnesses of its own selection and chosen only from the government employ, may assume elsewhere the aspect of an official procedure kept free from public clamor and from political influence, but it becomes here, in application to the present aeronautical problem, a star-chamber proceeding in which a great industry built up largely in response to government demand stands in danger of being so deemed without—contravention of the very fundamental of justice.

Misinformation is common, and properly so, about the effect of the coming appropriation bills upon the stability of their business and upon their capacity to keep intact at least the nucleus of carefully-trained organizations. It is not too much to say that the very fate of the American aircraft industry, as we have come to know it in the past three years, and its future usefulness as a national asset hang upon governmental action with respect to the appropriations bill for next year. That fate will be most largely decided not when the appropriations bill reach the President and become the subject of Congressional action, but before. On Oct. 2 and Nov. 25 in a pretty furnished office at the end of a long corridor in the Treasury building.

The path of the industry is plain and straightforward, if the government authorities will give it a proper opportunity to speak in its own defense. There is no occasion for any such detective-story maneuvers as the writers of political news love to describe. The industry needs no professional lobbyist to provide lavish entertainment for Congressmen and bludge the Washington sans guile. Such ploys are usually much more successful in backbiting their employers than in affecting legislation. The course for us is much simpler and much more effective. The aircraft industry requires only representation by its own leaders.

In the days when it was the fashion to have a new organization of the aircraft industry every spring and every fall, men who were active in the industry and who knew its problems at first hand were given their day in court. The names of Keys, Lawrence, Martin,

Locking, Yough, and P. G. Johnson are on the record among the witnesses. They testified, and they were cross-examined. It was a good practice. Let us hope that the high officials of several manufacturing companies and air transport lines will again be invited to appear in Washington and lay their cards upon the table. Let us hope, and express it in the form of an earnest plea to those who have the power to bring such an institution, that the industry's leaders may go before the Bureau of the Budget at its hearings, where the press is barred, and where no record is kept for subsequent publication, and where no lobbyist has the chance of penetrating. To Col. J. C. Rook, Director of the Budget, we address a solemn reminder that he, more than any other man except the President himself, has the immediate power of determining the future health of the airplane industry and its strength as a national asset. To Colonel Rook we put the direct question: *Will you give the men who have shaped the industry and who are best able to understand and foresee the effect of any change in its relations with the government, the chance to tell their story at your hearings?*

Flying the mail for twenty years

IN THE midst of the preparations to meet this issue of AVIATION to press, there occurs an anniversary of one of the two or three greatest events in the history of air transport. Exactly twenty years ago, on Sept. 23, 1911, Earle L. Oregonian flew his reconnaissance from Natick Boulevard to Minot on Long Island, a distance of five miles, and carried a sack of mail with which he loaded the postmaster who was waiting at the far end of the route to receive it.

The historians may argue over whether or not that was the first airplane mail experiment. Service veterans were tried as spectacular additions to aviation routes in the various European countries. In any case it was not literally the first useful delivery. For that one must go to the Balloons-Ponts which kept belated Paris in touch with the outside world during the war of 1870. But nevertheless the Oregonian flight was a precursor for the world, for the United States postal authorities have had these of all other nations in following up the experiment and putting it into practical commercial use.

It was the first faint and almost trifling beginning of a service which has built up a tradition seconded by none in the world. Tradition has grown and remained ever strengthened in the keeping of that dauntless bird, the air mail pilots of the United States—from Earle Oregonian to the newest recruit—none fazed and others unworried—many still in the service and many dead in line of duty,—the carriers whom neither wind, nor rain

and snow, nor gloom of night has stayed from the completion of their allotted task.

A long succession of postal officials have shown extraordinary vision, especially in the very earliest days, in recognizing the airplane as an instrument of postal transport. The general nature and immense scope of its services was forecast by the Postmaster-General at the time of the 1911 venture, and again by his successor in the Wilson administration. What they saw from twelve to twenty years ago has been but slowly appreciated elsewhere, and it is only very recently that air mail has been universally recognized as one of the brightest stars on the aeronautical horizon. In Europe it has been an incident, and a minor one of air transport, and only within the last few months have new specifications been issued for new planes especially for postal service. In the United States it has been the cornerstone of the building and the keystone of the arch, and in 1930, 75 per cent of the gross income of all the airlines in the United States came from the carriage of air mail matter.

We have often been exceedingly critical of the details of the policies of the Post Office Department and of the way in which the air mail service was being administered at the moment. No doubt we shall be critical again from time to time. That very fact makes to all the more eager to acknowledge the positively remarkable debt that American aviation owes to the Post Office, and especially to the distinguished line of far-seeing men who have served as Postmaster-General, Hitchcock, Burdett, Hays, New, and Brown, and to their right hands in moving the mail, Proctor, Shepley, Hendrickson, and Glover.

Financial depressions and airport development

ALMOST all of the innumerable panaceas for depression that have been bandied in the past year have had the common thought that it is better to subsidize work than to subsidize idleness, and that it is both socially and economically sounder to provide men with jobs than to pay them for not having them. Specifically, one of the most obvious measures of relief is through the expansion of public works. It is a measure that has particularly commended itself to the administration, and the President has never lost an occasion for explaining the federal government's endeavors to improve unemployment conditions through accelerating its building construction. The public works program as now being applied are subject to certain limitations in scope and usefulness. Aviation can, at least in part, remove these limitations.

The amount that can be wisely spent on public buildings is limited by the need for new construction

The equality with which the public works can be applied is limited by delays for planning, for developing the plan for a new building and getting ready to let the contracts in at best a slow process.

But there is at least one type of public construction to which these disadvantages scarcely apply. The need for airports is all year. There are many areas which should have fields and for which no provision has yet been made. There are many airports which are still but little advanced from the worst lot they could have and which are equally in need of grading, leveling-out, of runways, provision of new buildings, and the like. It is conservative to say that 80 million dollars could and should be put into airport development, without economic waste and without the expenditure of a nickel where it will be rendering a real service to aerial transportation, and it can all be done without aerial delay.

One of the virtues of airport work in the present emergency is that most of it requires no long consideration or elaborate plan. Surface work need not only upon an engineering survey of the ground to be broken. Buildings can be largely of standard type, and can be constructed for without long architectural preliminaries. Fifty million dollars worth of work, or thereabouts, will only upon the spot be set and the assurance that there will be money with which to pay the bills.

The first wave of enthusiasm for municipal airport construction subsided in 1920. To avoid new municipal activity there must be some new incentive. Local enthusiasm must be kindled up by the pressure of the general interest in an improved national airport network.

There is an obvious way of developing such an incentive, and also of insuring that the municipalities which have depleted their credit will have money to go ahead with needed work. Both can be accomplished through a system of state and federal aid. There are many cities which are today in such desperate financial straits that they are quite unable to take care of all their relief problems without external assistance. Assistance may be given in providing food and shelter for the unemployed, as many states are now preparing to give it, or it may be given in providing productive work, which is far better. The needs of cities is on the whole weak. That of the states is much stronger, and that of the federal government is entirely unmet. In one way or another the states are obliged to aid the weak. They can hardly do so more effectively than through measures which will at the same time provide relief for the needy and lay the foundation for the further expansion of the air transport map, and for a more general use of aircraft. The social policy which subsidizes airport work at the moment is in rendering direct and indirect service to all its citizens and a double service to some of them, jobs in the present and a public utility for their future

28 min. per hour and from Glen Corbin for Newark at about 40 min. per hour. Each leg is slightly more than 10 miles long; the fare is \$5 for the round trip and \$3 for leg fare. The service was well patronized during the first weeks of operation, especially as much by aerial sightseers as by those using it for recreation.

Robertson Airlines has introduced express service as its St. Louis-New Orleans route in cooperation with the Western Union. Passenger fares have been reduced, the through fare being \$11.50 (\$6.64 per mile) instead of \$17.85 (\$9.96 per mile).

Passenger lines Ogle concession

Pan American-Gulf has succeeded at last in obtaining permission from Ogle to carry passengers as well as mail over its route into Santiago from Peru. Passengers may now fly all the way from Miami to Santiago by Pan American instead of changing to a Chilean National Airways plane at Ayacucho.

Pan America has secured a substantial stock interest in Colombian Aerolineas, one of the most modern and one of the oldest (it was founded in 1922) airlines in the world. Passenger, mail and express services of the two lines will be closely coordinated. Aerolineas has 2,500 miles of airways, Pan American operates 10,600. There will be one P.A.A. representative on the Senate board of directors.

Lines win hearings

American Airways and Century Airlines have been granted certification of convenience and necessity for operating between Chicago and St. Louis. Their petitions were granted by the House Commerce Committee following hearings in May during which the Chicago & Alton Railroad objected strenuously to a grant on the ground that surface service is disruptive and not dependable. The protest was later withdrawn.

New York & Western Airways, operating between New York and Pittsburgh, has put on three additional Lockheed Vegas and Transcontinental & Western Air has bought seven Northrop Alphas for extension of its Kansas City-Los Angeles night and service route to New York City. Lindbergh has acquired two Lockheed Constellation for a 48-city, one-stop passenger service between New York and Washington, supplementing its hourly service. Two trips daily in each direction will be flown.

Imperial Airways plans

Imperial Airways has opened in the heart of London a new Airways Terminal Building which serves executive headquarters of the company and as city passenger station. Doors run direct from here to the Croydon Airport.

The company now hopes to open the Mwanza-Lake Town section of its African service in December, depending upon the delivery of the Handley Page H-4 (Humbly) type which it is to receive. The Handley-Heinkel machines, now on the Cairo-Khartoum route, for the new African section of 2,950 miles. A two-year reciprocal agreement has been closed with Geopos for the operation of Imperial Airways planes across Geopos Airways in both the African and Indian services, thus completing arrangements for operations in the Indian-Ocean area. One of the new Handleys was shipped down to Thailand early August when a propeller broke, damaging a second engine. In the landing the tail was sheared off by wires but the machine was not seriously seriously damaged and is now under repair.

The Argentine Air Transport company, one of the South American subsidiaries of Aeropostal, closed in third year with a loss of \$200,000, an increase over a loss of \$100,000 for the year before.

Air mail developments

The proceedings about the Post Office Department's methods and terms in granting air mail contracts has continued, mostly around criticism advanced by Lindbergh Lane, one of the strongest of the independent operators claiming they have been slighted. Lindbergh has announced it unsuccessfully offered to carry the mail between New York and Washington for \$625 per mile in comparison to the minimum of \$100 claimed by Eastern Air Transport. Lindbergh alleged interference with the operation of additional planes by Eastern (which Lane, Lindbergh's Central Office in Philadelphia) has caused the Post Office

Department to threaten to transfer and remove Lane from Central to the Philadelphia. Navy Yard field Representative Clyde Kelly of Philadelphia, member of the House Post Office Committee and sponsor of the original contract for mail bill, has declared he will instigate a Congressional investigation of the Post Office methods if the inquiry already provided for in the House Appropriations committee does not satisfy him. Master Air Transport has joined the independent in their fight against present methods of awarding contracts.

New ship-to-shore mail record

Impressive demonstrations of ship-to-shore air mail deliveries were made in August by the Bremen's plane and in September by the Europa's machine.

The Bremen plane left the ship when it was 700 miles from Boston on Aug. 10 and landed in Boston harbor at 6 that afternoon. Its mail was received in New York City the next morning, while the Bremen was not due until late the afternoon of Aug. 15.

The Europa's machine surpassed this performance by leaving the ship on September 12, while it was 600 miles from New York. After leaving at 5:45 that afternoon, the plane was delayed and flown to Bridgeport, Conn., where it was fuelled again. It arrived at the pier in Brooklyn 22 hours after leaving the ship and about 18 hours before the steamer was due to dock.

Canada adjusts mail situation

The routes to be dropped from the Canadian air mail system have been selected and the reduction is set to begin.



FIRST IN CIRCUIT OF ITALY

Support: Canadian aircraft built the first 100 miles by air in victory in the second round of first prize race around Italy.

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October, 1932

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October, 1932



10th Post

MORE MANEUVERS

After 100 planes of the United States military air base parked during the recent annual exercises in Europe, maneuvers in northern Italy.

where were an extreme to yet, expected when it first landed late last spring. On Aug. 19—between two visits reported the temporary one-way traffic. Canadian Airplane No. 1, had been operating since June—only two other services were dropped, making a total of five. The first two are the Montreal-Toronto service and the Montreal-Hatfield-North Hatfield-Lancaster of the private service. The three previous aircraft were the Montreal-Montreal-Saint John, N. B., Montreal-Quebec and Toronto-Hatfield lines.

These dropped lines were the last heavily patronized lines, not considered part of the trunk system. Among those retained are the Montreal-Buffalo service which connects with Canadian Atlantic Ocean and the international services under this company. Edmonton now 300 miles out of the U. S. service as Alaska is credited as by way of Calgary only. The only passenger line is reduced to 75 miles per mile on the Toronto-Toronto route.

Canadian Airlines officials are now improving the Avia 627 mail plane especially designed in England for Canadian operators. The machine looks much like a typical Douglas or Stinson mail biplane. It is fitted with an Armstrong Siddeley Jaguar Major engine of 331 hp. It is equipped with a Townsend ring, cranes at about 140 mph, and with a fuel tank of 200 to 300 gallons.

More radio range becomes

Amplification of the federal airways system during the present fiscal year will include construction of 30 radio range

stations and completion of fifteen beacons now under construction. Fifty-one are in operation already. To the 45 radio communication stations now in operation there will be added eleven now under construction and ten which are planned.

The new route of the New York-Washington airway has been selected and routing between will be completed soon. It follows much the same course as the old airway but has Newark as its northern terminus. Service was to be begun about Oct. 1 on the proposed inland airway between Louisville and Dallas by way of Nashville, Memphis, Little Rock and Tulsa. It probably will be made for night operation by early spring.

The third of a proposed series of 92 sectional ranges, controlled by the Department of Commerce as maintained as the contemporary spot map and likely to eventually supersede that type, has been published. Known as Type R-7, it covers the Detroit-Pittsburgh region between 2 deg. of latitude and 6 deg. at longitude, or a territory roughly 220 miles east to west and 150 miles north to south. Previous section maps are centered about Chicago and Milwaukee.

Airport affairs

The Department of Commerce has issued a new bulletin designed as Aeronautics Bulletin No. 2 on the subject of design and construction of Canadian operators. It is composed of the National data on trends in airport construction with a list of projects in progress. It is placed on the shelf for

better coordination of airport construction plans with work at other branches of the industry in the future. Road building and construction of other air construction only a remote possibility.

The mid epidemic at airport last spring has been followed by a better record. Airport's most serious loss occurred at Lambert Airport, St. Louis, where eight constables, American and German, including most damaged to the extent of about \$11,110. Delicate wiring in the hangar was given as a probable cause.

The new building program in field range field, which will include new hangars, an administration building, officers' and enlisted men's quarters, is well under way. National Air Transport's new \$125,000 hangar and office building in Kansas City (Mo.) Municipal Airport has been completed. The \$14,000 communication building at Oklahoma City Municipal Airport should be finished by early fall. Curtiss-Wright Flying Service closed as Miami Field, Indianapolis, services with the sale of its larger, built at a cost of \$14,000, to the Indian National General Aviation squadron for \$25,000.

Personnel changes

Rod G. Egan, vice president and general manager of American Airlines, has been named president of the University of the South, where he has been working in the development of air line, among those who have been named members of the United States Aeronautics Commission.

C. L. Egbert, formerly chief engineer, has been named president of the Boeing Airplane Company. G. M. Carr has been made plant manager.

C. E. Ruff, formerly treasurer of the United Aircraft Company, now is treasurer of United Air Lines, and H. H. Bowman is treasurer of United Aircraft Company. Ruff, formerly treasurer of United Aircraft Company, now is treasurer of United Aircraft Company.

Anna Barker has been appointed State Commissioner of Aviation in Alabama.

Roy E. Osborne has been appointed state director of airports in Kansas. He is a graduate of the University of the South, where he has been made president of the Curtiss-Wright Flying Service, as well as president of the Curtiss-Wright Flying Service, as well as president of the Curtiss-Wright Flying Service.

Walter has resigned from the American Civil Service Commission. J. A. Alford, for more than three years vice-president of the Curtiss-Wright Flying Service, has been made president of the Curtiss-Wright Flying Service, as well as president of the Curtiss-Wright Flying Service, as well as president of the Curtiss-Wright Flying Service.

matter of how annoying is one of the two outstanding problems for next year's organization. It should receive serious consideration long before the race.

Score-boards at last!

In one respect the announcer's problem was simplified because number of winners, for the first time in nine years, were scored on a hand for four spectators. For anyone who wanted them, official programs were available with complete lists of the names of the race's host and of their places and power plants. That alone would suffice to establish the Cleveland race's place in history.

For the first time since the Earhart-Bird first time, the spectators were seated in the middle of the race course and most of the race took place behind their backs. The experiment was tried to avoid conflict with commercial airlines, leaving half the airport completely clear of interference by the race and in the opposite interest of safety for the racing pilots and spectators. That alone would have been a great achievement. Additionally there was great danger where it is Chicago the pilot who continues his course past the home pylons without making the turn will go straight into the grandstands, but as Cleveland no such grandstands existed. If anything there is more danger in turning towards the crowd so at Cleveland, this is making a turn away from it, while the latter type is far more exciting, it is much less dangerous. The avoidance of interference with commercial traffic is a delicate problem but fortunately the procedure, used at Cleveland in each take-off, was positively always made across the field from left to right as seen from the grandstand. It would then be possible to leave the necessary part of the field clear for commercial traffic by turning the lower pylons well to the left and making a very steep turn (about 120 deg.) at that point. The course would be far more interesting to the crowd in that case, and there should be no interference danger for the pilots. A one grand layout is shown in the sketch on page 205.

The standing start, with all the machines going together at the dropping of the same flag, again badly confused them. One unfortunate feature of the standing start business, was that all pilots were taken from the starter's flag and with the calculated speed based on the normal distance across the course they were to be expected to have making no allowance for the time lost in the standing start as for the distance of about 1/4 mile to the starting pylons and back. In such an advantageous as a standing race the pilots actually had to fly 3/4 miles plus a standing start, but the speeds announced for such a race were in all cases about 8 to 10

mph, below the true speed of the competing machine around a closed circuit. Last year the course was unfortunately shortened to about 1/2 the effect of this year. This year it was unfortunately lengthened to allow for the starting pylons, with result almost equally unfortunate. The time of a race starting from a stand should in all cases be taken when the first machine crosses the starting line, for the beginning of the race. The time of a race starting on wheels will then have some definite significance in terms of the performance of the airplane. For example, the average speed announced for Boylston winning the Thompson Trophy was 236.25 mph, yet he actually flew five of the laps at over 340 mph, and his average for the last one lap, all those not directly affected by the standing start was over 229.

Straightaway speed trials

In the endeavor to interest interest in the racing a great number of "straightaway speed dashes" over a one-mile course had been planned for this year. Unfortunately the conditions were not so favorable. The straightaway races proved to be a total failure. They were much more interesting with planes as fast as Boylston or Donaldson's.

The course there was shorter than in past years and so that respect was an improvement, but they are still too long. As a general rule a race is divided in the first couple of laps. It would be more interesting to run a race over a course of one or two or three laps. Perhaps allowing permitting the whole race to be finished in about seven or eight minutes, in the ideal length for all events except those for the very distant machines.

The handling of the field and of the controls was, in some ways, the worst of the race. Cleveland. This session handicapped with an usual audience.

Handicap races and amateur pilots

The scheduling of the race has been mentioned as one of the two most serious weaknesses of Cleveland. The race was the strongest feature in providing for handicap races. Nothing the organizers did in potential meant to the crowd, and even to the competitors. They have been neglected too long. There can be no possible excuse for failing to put them on the program next year. There is no space here to discuss in full detail the way in which they ought to be organized and the handicaps assigned, but that and other matters pertaining to the detail of racing rules will be covered in another issue. It is our hope to be able to cover it as a separate race the pilots actually had to fly 3/4 miles plus a standing start, but the speeds announced for such a race were in all cases about 8 to 10

apart from a down electrical machine of some medium-powered sport or training type and few flies in a series of airplanes from the best known pilots with the machines being assigned by lot in each heat. Although it may be that some of the pilots would have been unwilling to participate, there would have been particular possibilities as a popular attraction in a series of such races among the foreign pilots at the meet and some chosen American representatives.

At Chicago in 1930 the sportsman pilot race, first attempt and then a second, proved a dismal disappointment. At Cleveland elaborate preparations were made for a group of amateur pilot events, but they ultimately disappeared from the program entirely, except for a single rather unsatisfactory race on the last day. Starting competition in the air seems to be in a bad way. There are several reasons, but one of the most important was suggested in the reception given the sportsman pilot's "Derby" that same year. Cleveland during one afternoon's racing. It occurred only the most casual attention from the spectators and the newspapers when it arrived, but the way had been nicely prepared for it by the local companies which had habitually catered to it previously, not without inspiration from someone engaged in aviation, by such endeavoring planes as "Smiley Derby," "Blue Bird Derby," and "Blackbird Derby." So long as it is recognized pilot whether or not he is engaged in competition, continues to be an object of interest, at least so long as the aircraft industry continues to wonder what is wrong with the private plane market.

In one detail of general organization the Cleveland race was definitely inferior to that at Chicago. There was not enough contribution of field control to the race. Each event should have a permanent place where he can overlook everything and equipped with field a down telephone and radio phone and a complete master control enabling him to act in anywhere on anything. His is the function of general control and of instant action in emergency. At Chicago Major Schuster occupied that post. At Cleveland there was no one in that position for long. Since just the experienced level of the grandstand structure should establish he set aside for that purpose.

It is a fact that almost immediately in speaking of events managed by Cliff Henderson, do say that the general management was close to perfection. The usual remarkable feature of the race and having personnel were accomplished. Gathers and when more, as noted, with interest first what first race was a success and then what second was a failure. The provision for the press was excellent and for that speed airplane is due to James H. Loepp, Herman T. Adams, and John H. Neill.

A duty and an opportunity that seems to have been overlooked by the several states of this country. It is to contribute to the development of extensive private and transport operation of aircraft in anticipation of the future tax revenue to be derived from these sources. The federal government has done its share but there are legal limitations beyond which it cannot proceed. Some states have carried on from these limits. It is thus that the others give some consideration to the problem.

The opportunity of the states

By Charles L. Lawrence

President, Associated Chamber of Commerce of America, Inc.

WHEN all business was on the loose in 1928 and aviation was started with it, numerous airplane factories were built and thousands of planes produced in the belief that aviation was simply a new form of motorized speed on the air.

When the general business boom subsided, and aviation diminished with it, there was a tendency to go to the other extreme and regard flying as a long distance transport analogous to the railroad form of steamship traffic, rather than the automobile.

Neither view was correct, and I feel warranted in now declaring, even when pessimism is popular, that aviation is rapidly approaching the time when the airplane will not only serve as a basic common carrier for the public but also as personal transportation for thousands of people who demand speed and are willing to pay for it.

The air mail and transport system, which today flows the backbone of commercial aviation, while of course performing a valued service between cities within the same state and a considerable distance apart is essentially an intercity operation. The quality aircraft production of the future will come as a result of the development of flying privately and for pleasure in character.



It is likewise true that private or individual flying will appear as a major factor in social and business life, when—and only when—the aerial states recognize their opportunities and responsibilities. When extensive private flying will mean to the aircraft industry as a market and to the interstate air mail and transport system as a feeder, as indicated by the fact that in 1929—a significant year—2,500,000 people on private or charter service, actually within state boundaries, flew, it is actually approximately 136,000,000 miles and covered 3,000,000 persons. The opportunity of the states is definitely to encourage this phase of aeronautics. Their responsibility is to do so in such a way as to be equitable. The demand and, I be-

lieve, can best be achieved through the application of uniform rules and regulations in harmony with those promulgated by the federal government, and as the construction and maintenance of airports and aids to navigation require state funds.

Little has been done thus far by the states as a group to encourage the development of aviation, even for their own advantage. Legislation governing the regulation of aircraft within their borders was enacted by a number of states long before the Air Commerce Act of 1926 went into effect. Today, there is not a state in the Union that does not attempt to regulate, or, worse, stifle activity in one way or another. Many have passed enabling acts, making it possible for counties and cities to acquire property and operate airports, but little from the enforcement of such agencies is speed money, the support afforded by aviation by the state has been largely negative. For the most part, they have been content to sit by, watching the federal government carry out its development program. Because of this, aviation has suffered and the states have deprived themselves of a constructive factor in its growth.

Before it is too late for the situation, let me quote from the statement of policy which it was my privilege to

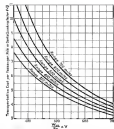


Fig. 2. Transmission cost per passenger-mile (left) is the ratio of an operating cost per hp-hr to 46 cents. Note that dead weight transmission cost actually varies almost linearly to increase of actual operating cost per hp-hr (see Fig. 1).

A few examples will serve to illustrate the use of Fig. 1. Consider a single-engine plane of 400 hp, which sells for \$10,000, that is, for \$40 per rated hp. Under the above assumptions, take the value for operating cost (dead from Fig. 1). This is found to be 6.60 cents per hp-hr, which means that the operating cost of this plane will be \$27.20 per hour. If it has a cruising speed of 120 m.p.h., the operating cost per mile is 22.67 cents. If a two-engine plane of 1,200 hp costs \$28,000, the cost per rated hp is \$23.33, and the operating cost is 5.925 cents per hp-hr, or \$69.30 per hour.

Since the operating cost factor has been explained, it is possible to proceed with the comparison of the remaining part of Equation 5. If the weight of powerplant and fuselage is taken to 200 lb, the factor R of specific fuel consumption at cruising speed is 0.5 and average terminal speed is found to be 100 (which means that all available seats are used and paid for). Equation 5 becomes

$$C = 5.66 + \frac{250}{\left(\frac{W}{200} + R\right)} \quad (7)$$

In this equation all factors are known except two. The factor R is usually determined by the weight of the operating unit and is one of the characteristics of the transmission which reduction factor to a very large degree. The other factor can and must be influenced by the operator, depending on the term $\frac{W}{200}$, which has been here termed "the transportation coefficient." It is the only factor in this formula which the designer can alter materially by increasing the efficiency of the machine

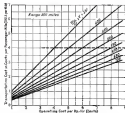


Fig. 3. Transmission cost per passenger-mile for various values of transportation coefficient and dead operating cost

as far as the weight and the aerodynamic qualities are concerned. Since this factor is on the lower side of the factor, it must be increased to decrease the transportation cost. This can be achieved (1) by decreasing the weight of the engine, but fully equipped airplanes, (2) by increasing the gross weight and thereby the useful load, (3) by increasing the existing speed without increasing the engine load, or (4) by decreasing engine power without change in cruise and cruising speed. All these methods can, of course, be employed to increase the transportation factor.

To obtain an average value the transportation factor has been calculated for a number of typical transport planes. This figure and typical results have been taken from a comparison of the results in Table 1.

According to the above definition, the term R is used in calculating the transportation factor. The use of actual payload and fuel

load. Weight of crew and balancing act is not considered part of useful load, but as part of the weight of the plane. If radio, or other special equipment is needed, its weight has to be considered as dead weight also and has to be deducted from either the fuel or payload.

Table VI shows the values of transportation cost per passenger-mile in cents, as related to transportation factor R .

Table VII shows the values of transportation cost per passenger-mile and the range of the airplane. The value for operating cost per hp-hr used in this table is 50 cents. This makes it possible to take the values of transportation cost from this table or from the curves of Fig. 2, and to multiply them by one-fourth of the actual operating cost per hp-hr (which can be taken from Fig. 1) for the specific machine under investigation. It is then annual transportation cost per passenger-mile for any given range. The transportation factor has to be calculated of course, from available information, taking care to use only

Table 7. Values of transportation factor for typical transport planes
500 mile range at cruise speed
Distances are based on those in aircraft's operating manuals

	Boeing 347-20	Boeing 347-30	Boeing 347-40	Boeing 347-50	Boeing 347-60	Boeing 347-70	Boeing 347-80	Boeing 347-90	Boeing 347-100
Weight	1200	1400	1600	1800	2000	2200	2400	2600	2800
Operating cost	112	128	144	160	176	192	208	224	240
Operating cost	100	110	120	130	140	150	160	170	180
Weight	1200	1400	1600	1800	2000	2200	2400	2600	2800
Operating cost	112	128	144	160	176	192	208	224	240
Operating cost	100	110	120	130	140	150	160	170	180
Weight	1200	1400	1600	1800	2000	2200	2400	2600	2800
Operating cost	112	128	144	160	176	192	208	224	240
Operating cost	100	110	120	130	140	150	160	170	180
Weight	1200	1400	1600	1800	2000	2200	2400	2600	2800
Operating cost	112	128	144	160	176	192	208	224	240
Operating cost	100	110	120	130	140	150	160	170	180

Fig. 4. W. H. H.

Table 8. Values of transportation factor for typical transport planes

each value for R , as is explained in previous paragraphs.

The loading factor R is in both Table VI and Fig. 2 is assumed to be unity, that is, all seats are used and paid for. The weight of each passenger and his baggage is taken at 200 lb.

Fig. 2 shows some exceedingly increasing facts which have great importance for commercial aviation. For instance, for visualization of the interrelation between aerodynamic and weight characteristics of the plane, the range as reported by the airline operator, and the transportation cost per passenger-mile. It can be seen that only close coordination of all factors will bring low transportation cost.

Range has a greater influence on transportation cost than is usually recognized, especially for planes of low engine efficiency, that is, for those with a low transportation factor. An aircraft of range from 300 to 600 miles requires transportation cost per passenger-mile of 1.51 per cent, for a machine with a transportation factor of 400, and by 41 per cent for one with a factor of 40. Percentages for other cases can be easily figured from Fig. 2. For example, airline operation a range of 400 miles between stopping points should be much improved. To have the necessary range of fuel to the cost of fuel must be obtained on account of weather conditions. The same distance flown between stopping points should then not be twice the cost.

A practical example

A sample calculation of operation cost, transportation cost and factor, showing the influence of range, for a typical airplane equipped with an engine of 420 hp, is shown in Table 8. The weight is 2,200 lb., its gross weight 4,800 lb., and its cruising speed 140 m.p.h. The weight of pilot is 170 lb., that of the laboratory of 90 lb., and the weight of miscellaneous equipment a 120 lb. The model load (as explained in this study) is therefore 1,850 lb., which includes actual available payload and fuel load at take-off. The transportation factor

$$R = \frac{W}{200} + R = 12.5$$

If we assume for this plane a rate point of \$10,000, or a basic sales price of \$25.00 per rated hp, the operating cost per hp-hr is found to be 6.605 cents. From Fig. 2 the transportation cost per passenger-mile is found by multiplying the value of estimate of

$$R = \frac{W}{200} + R = 12.5$$

by 0.665. This gives the following results: range 300 miles—transportation

Table 9. Transportation cost per passenger-mile
Calculated with average cost per hp-hr of 46 cents

	Boeing 347-20	Boeing 347-30	Boeing 347-40	Boeing 347-50	Boeing 347-60	Boeing 347-70	Boeing 347-80	Boeing 347-90	Boeing 347-100
Weight	1200	1400	1600	1800	2000	2200	2400	2600	2800
Operating cost	112	128	144	160	176	192	208	224	240
Operating cost	100	110	120	130	140	150	160	170	180
Weight	1200	1400	1600	1800	2000	2200	2400	2600	2800
Operating cost	112	128	144	160	176	192	208	224	240
Operating cost	100	110	120	130	140	150	160	170	180

cost per passenger-mile 296 cents; range 400 miles—335 cents; range 500 miles—342 cents; range 600 miles—348 cents.

Aircraft now an increase of cruising speed at this plane by 30 per cent to 154 m.p.h. without changing hp or weight. The transportation factor be-

$$R = \frac{W}{200} + R = 12.5$$

comes 12.5. This gives the following results: range 300 miles—transportation cost 296 cents; range 400 miles—335 cents; range 500 miles—342 cents; range 600 miles—348 cents. The effect of reduction of weight of the entire plane, of increase of gross weight, of increase or reduction of engine power, or of first cost can be evaluated in similar manner. This again for needed very useful for determining whether increase of speed, obtained at increased cost, is worth it or not. It is usually not possible to obtain a satisfactory answer.

From Fig. 2 shows the value of

A German light plane engine

An inverted four-cylinder, air-cooled, light engine for light aircraft has been designed and built by H. Hirth, which is the only of its kind in Germany. The engine weighs in total

weight 250 lb. and is available in two designs: 100 hp and 125 hp. Both are available in two versions: 100 hp and 125 hp. The combination of all factors is said to result in exceptional smoothness at operation at all engine speeds.



The Hirth engine



By Douglas W. Clephane

Selling the business plane

The present depression is forcing executives in all branches of business to a realization of the need for closer contact in these fields of activity. This can be accomplished most efficiently by airplane. In the accompanying article the author discusses this and other phases of the business market and analyzes some three hundred sales of airplanes made to business men.

AS BUSINESS enters the upgrade there will be an intense effort on the part of all airplane manufacturers, and dealers and distributors, to reach the large potential market for planes among non-aeronautical corporations. Over 300 planes have already gone to such buyers, but the amount of business expected has failed to develop, not only because of the general business depression, but because the industry as a whole has not been able to sell the fact that ownership of an airplane by a large company will result in a very definite time and money saving, a tremendous increase in productivity, and an indirect advertising and publicity value.

The coming months after a quiet pause for opportunity for well-directed sales efforts in this direction, for during the last year numerous of the largest companies, from the gasoline oil down, have been out in the field more than ever before, and are ripe to consider any proposal that will permit more intensive field contacts over a wider area.

Before summarizing the experience of the writer in hundreds of contacts with corporations themselves in selling privately owned airplane transportation, a breakdown of an incomplete list of 300 companies who have purchased their own planes should prove illuminating. Despite the inadequacy of available in-

formation, the percentages on page 590 show the fields in which the companies owned airplanes has grown of greatest value, and the type of companies that own the largest number of planes.

The oil companies have purchased by far the largest number of planes, when checked by industries, and only a small proportion of them are most difficult in procuring the sale of aviation gasoline. The airplane has afforded greatest usefulness in oil-making surveys, transporting crews and equipment parts, and general transportation service. Next in volume come banks, paper, mining, power and other corporations engaged in making new materials available, and requiring surveys of territory hard to reach by ground transportation.

Probably the greatest potential market for planes is among the thousands of companies that do a large volume of

business in small units requiring thousands of small orders, tremendous expenditures to keep the name of the product before the public, and close contact with bulk sources of supply and the distributing organization.

The classification includes companies making and selling food products, cosmetics, chewing gum, medicine and drug supplies, paint, oil, other supplies, paint, clothing, and other such products. This field has absorbed about a tenth of the private planes now being sold partly for business, and is one in which the airplane has proven of greatest value.

Dependable, prompt, expensive machinery, or having need of frequent personal contact with, and service work, by highly paid specialists, have found the privately owned airplane very useful as essential equipment. These organizations such as General Electric and American Telephone and Telegraph have purchased their own planes both for representative purposes, and for general servicing transportation use. A few chain stores with their thousands of sales in all parts of the country have experienced with the use of airplanes,

but the use thereof has not been as widespread as so successful it would be expected.

Quite a number of airports recognize their own planes, and many more are experimenting with air delivery of papers to nearby towns in chartered planes. A few publishing firms have issued the ownership of their own planes a profitable investment, but the number of large organizations of this character is quite limited. A few banks, insurance companies, and other companies have purchased their own planes, for use in transporting money and accounts, sales work, and contacts with important clients, but the type are steadily increasing a long enough distance to show a great saving in time, and it is expected that regular routes will be established by the airlines for the transport of money and securities.

Another of the largest potential fields for business sales is professional salesmen, such as lawyers, doctors, writers, architects, and engineers who have to make frequent long business trips.

Future purchases

A detailed survey of future sales of planes for business use leads to the forecast that corporations in the above listed fields will purchase nearly all the airplanes bought for private use in the airplane manufacturing industry for some time to come. The dealer and distributor have made little attempt at a detailed market analysis of the possibilities of sales for business use, and the output out of a detailed sales program to reach each type of business, such as would have been employed in any other industry.

Having determined through past experience where our market lies, the next extent to make the sale program is how to reach it. Bank of the twelve listed fields will require a different type of approach, and different data. Lists can be made up from trade directories and many other sources, or supplied by the houses, of the largest companies in each field. In general the larger corporations are the best prospects, although they must be taken care to overlook the smaller, but sometimes more progressive organizations.

One of the most surprising phases to note in selling business transportation, is the fact that consumers of practically all the largest corporations will answer any personal letter concerning the purchase of a plane. They are immediately inclined to be intrigued for such a recommendation they jump at the chance to obtain accurate information, and

when will go into detail by mail concerning their transportation problems, and what they would like to know.

Personal procedure

Among determined who the individual prospects are, propositions must be made to approach each one of them in a manner to bring results. In the past many attempts have struggled to sell airplanes as they would a diamond, giving it without knowledge of the proper points to use, with an air of formality as to the necessities of the company, or how it might employ a plane. Too many times when sales of such a kind declared he was not interested, the matter ended there. From concerning just suggestions sales prove that a person in charge of airplane acquisition from the time the matter of a plane purchase is first brought up to the actual signing of an order. Many sales have been made after contact over a period of from three to five years, but with a product selling for from \$5,000 to \$20,000 a comparatively large sales expenditure is warranted.

Having decided that a certain company could use a plane, the sales office should make a definite plan of procedure for each individual prospect. The first step is to find out who within the organization is concerned in ac-

quisition. Possibly this party will be the assistant sales manager, assistant to the vice-president, the son of some official, or a senior executive, who has no authority to make an actual purchase. When made a person at hand, he will often work as hard as the salesman himself to bring about the purchase. From him, detailed information can be obtained as to who has the authority to recommend the purchase, information as to who travels, where and how much, and general data as to the internal organization of the company, and the objectives of various executives to flying. This is an essential part of any sale, and without such very essential data the approach made to this information is wasted.

Corporations executives tell us that, while they have been approached many times to buy a plane, it is almost impossible to obtain any accurate information as to business. The average man wants to know if flying is safe, whether his insurance would be paid if he were killed, or injured in an airplane accident, how much it costs to operate a plane, whether he could land at his summer home, why some planes have one engine and others have two, how it would take him to travel his branch office in an hour with an airplane at his disposal, and how other



Manufacturers of machinery and heavy construction have found the airplane almost a necessity to enable high-located engineers and when officials in making the many contacts required in all parts of the country. The Douglas and Ryan Company has recently purchased its latest Lockheed.



J. W. McArthur, general manager of the French-Hill Battery Company has installed over 7000 miles in his plane, visiting dealers and all parts of the nation while representing

comparisons with an organization similar to be one their planes.

The salesman should devote his attention to the first ends on a pre-flight inspection. He should first be seen as a plane, and leave comparisons between different makes and types till later. He should be prepared with charts, checked, comparative material and airplane form between principal parts, written information sheets on cost of operation, detailed data on how other companies are using their planes with statements of their value from high officials, all available information concerning the safety of flying, and also on the effect of flying on personal insurance, good interior and exterior photographs, possibly showing permanent business men with their airplanes, and lastly should be prepared with a plan of operation for the particular prospect. The sales effort should immediately follow the salesman's visit with a letter thanking the executive for the interview, and taking up points in which the prospect is interested.

Selling via a comparative rate is offered by selling one man. Usually the president, vice-president, sales and marketing managers, the entire board of directors, and to general all those who might use a plane have to be convinced of its value and safety before an order is placed. Five companies will demand that an employee fly if he is discharged to do so or if he loses future fringe benefits. The salesman should not be afraid to try to do so much as a single call, realizing that from five to ten personal contacts with each executive will be required before they will even give serious consideration to such a purchase.

An advertising prospect

Advertising plans an important part in a corporate plan. While few direct inquiries may be received as a result, the executive will believe more in the safety and reliability of a plane when he compares one, those made by a company that relies almost entirely on personal contacts. The salesman has been able to reach three tiers of acquaintance places in a three-month period directly to trade paper and business magazines, advertising.

It must be emphasized that the danger of flying will usually be the executive's greatest objection to the purchase of a plane. Often this will not be expressed, but the salesman must always tactfully present all available information on the safety of flying. The families of executives often furnish the strongest block to the consideration of a sale, but that is another area resistance that will seldom be expressed. Every effort should be made to ensure that it is the parent and parent of a company that flying is safe, and they should be informed of the demonstrations.

Many a sale has been lost through

not using proper technique in giving demonstrations. The plane should always be freshly cleaned both inside and out, it should be warmed up before the arrival of the prospect at the airport, and only the most conservative flying should be done during such a demonstration. Many pilots attempt to show the standing qualities of their machine, but this will only increase the fear resistance of the prospect unless he has a great deal of experience in flying. A warm day should be selected if possible, and the prospect will always remember a trip over his home or factory. The next procedure is to illustrate it is to take the prospect on one of his regular business trips. This time money and many dollars will not do it. The prospect will be convinced that he is really to fly, but it is well worth while. Often the younger couple runs within the expectation itself, will do the salesman all where the president has a crash trip, or has an urgent shipment of materials to make. The writer has seen four sales close at a first visit of the salesman taking a first customer on a trip lasting from four to six

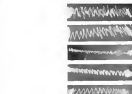
points and sometimes where planes crashed up due to faulty construction or design, he is doing the best thing in the world to tell any tale. The prospect will feel that if there are problems, he would rather wait until airplane design or general has reached the point where all planes are safe.

Making the executive choose

We have on every side that a pilot usually has the final say in the selection of a plane, and that communication is often necessary to outside parties to bring about the final order. This has unfortunately been true, but is largely the result of too technical a discussion with the executive, and of not getting him to understand comparative points himself. The writer has found that when the above outlined procedure is followed, the prospect usually feels that he is a capable judge, and will be very glad to choose for himself and associates.

There are many ways to offer services with which an executive is employed in other types of sales work. The aviation industry has not as yet seen it. It would be well worth while to copy the corporate salesman with a portable sales picture presentation machine, with specially prepared films to make a graphic presentation of construction features, safety information, comparative charts, use of planes in business, transportation of merchandise, and other points which can only be brought to the executive in a dramatic way. Copies of brochures, addresses, or letters from companies using planes, will be read with great interest by the aviation executive. Samples of advertisements can be mailed with all literature, and finally the salesman, whether connected with the factory or dealer, should have a complete sales manual showing photographs, and giving accounts that are references on any point which the prospect may bring up.

Last year all executives were engaged in close contact with all parts of their organization, and have learned that more from their contacts are maintained by the high industrial executives from the lower office. Lack of time is usually the excuse given for doing most of this work. Corporate sales are ready to be above how the airplane can aid them in getting business, giving sales, delivering their shipments, and general transportation, but the ground must be prepared first, and definite facts must be furnished, and the field sales given proper information work with, and trained in approach to this matter. The manufacturer or dealer who goes after the market with only a few salesmen, and no sales promotion, and a sufficient expenditure of money for direct results, should not have a complete of future business.



Illustrated engine operation chart descriptive to first phases, good, standard and delivery fuel.

THE Cleveland section of the Society of Automotive Engineers provided for only two papers on other than power plant subjects. These two, however, proved to be as colorful and as packed with illustrations and demanded such careful presentation that, even though the machine contained and most of the material was for only a small part of the discussion that members of the audience wished to continue.

The first presentation, for which Stephen J. Zied of the Pioneer Instrument Company was responsible, was entitled "A Study of Airplane and Instrument Board Vibrations." It presented the record of Mr. Zied's work in developing an instrument for measuring the vibration of any point in an airplane structure and its subsequent application of that instrument to an analysis of the vibrations of many points in a great number of airplanes, especially of the instrument board—the panel being based on a total of 216 test points.

Mr. Zied pointed out, what nearly all generally realized, that a very serious error is the neglect of instrument

vibration. It is given totally false information when assumed on a vibrating board. Nothing could be simpler, he stressed, than the fact that the indicator consisting of a steel ball rolling in a curved hole filled with liquid, but when a group of nine instruments of slightly varying form were mounted on a board that was hand vibrated with an amplitude of only 0.03 in. their readings fluctuated widely. Some of them continued to tell the truth, but others, in which the ball had too much or too little clearance in the tube, reflected much of the ball's varying from the truth by as much as 10 per cent. Similarly, when accelerometers, the parts of which had not been exactly balanced, were mounted on a similar vibrating stand, the results fluctuated so widely as to be completely unusable.

Instrument board vibrations

Mr. Zied pointed out that most vibratory trouble starts from the engine, that the greatest error must be taken to avoid resonance of the natural periods of instrument parts of the structure and the periods forced by engine rotation, and that the engine mount and airplane structure should provide for the largest possible amount of damping. Flexible engine mounts, such as those incorporating rubber, serve the dual purpose of isolating the vibration and insulating the natural periods so that resonance will appear at speeds below the normal speed of engine operation. The author had found, however, a marked difference of opinion on the usefulness of flexible mounts, the industry being about equally divided between those who thought that vibration could be reduced by making mounts more rigid and those who wanted to make them as rigid as possible. He pointed out that a very serious trouble seriously starts at the engine

the engine as by an means which is responsible. Identification is a controlling factor.

The vibrograph developed by Mr. Zied and his associates has the general appearance and use of an altimeter and contains essentially of a free pendulum, the motion of which is optically recorded on moving picture film.

Tests on airplanes showed that in practically all cases the vibration of the instrument board is as almost pure sine wave motion, in a very low frequency, the natural period of the board having nothing to do with it. There was exception to this, however, where the vibration was measured at a point very remote from the engine, the frequency in the rear cockpit of a large plane being found to be usually positively constant at 25 per cent to all engine speeds.

As might have been expected, airplanes varied greatly in response, as in magnitude of vibration. Mr. Zied considered any amplitude up to 0.01 in. satisfactory amount, and that standard was attained by them half of the airplanes tested.

The magnitude of vibration of course varies widely with engine speed, as shown in the curve reproduced here. With 20 per cent of the test is sufficiently extended times are of course two means in speed corresponding in the primary and secondary harmonics, respectively, the latter being at one-half

the g of the engine. In most cases the maximum amplitude does not exceed twice the minimum, but occasionally, when there is genuine resonance between the natural periods of several parts of the structure, the ratio may run much higher than that. In one case (the g is reproduced herewith), the vibration was so bad at 1,500 r.p.m. that the pilot was unwilling to fly the plane at that engine speed even long enough to make a measurement. Rotation of the screw through the points taken past above and past below critical speed, however, suggested that the amplitude at either of the maximum loads would have been at least 0.05-in. Vibration amplitude above 0.05-in. were classified by the 22nd as "extremely rough," causing headaches to the passengers and making the pointers of most self-inductance instruments oscillate 1 in. or more.

Another interesting series of experiments showed, quite surprisingly, that the vibration characteristics appear with age up to a certain point. In one case a low-engine machine showed a maximum amplitude 26 per cent higher than that of one of the same type after 50 hours service. In another case there was an almost identical difference between the stresses for a new machine and for the same plane after 300 hours of flying.

As the results of these experiments the author suggested the determination of his company to avoid any natural vibration or any natural frequency at between 1,000 and 2,000 r.p.m. in all aeronautical designs in future.

In the course of the discussion, Mr. Deane of British Air Transport recommended suspension of movement beyond the rubber shock absorbers in some cases. Not only had his company found that



FIGURE 1. Vibration curves of varied form for a low airplane.

an advisable means of reducing fatigue risk involves due to vibration, he said, but even in super cranker airplanes so mounted have sometimes been damaged when hardly anything else was left of the plane.

Load factors rationalized

Prof. Joseph S. Newell of the Massachusetts Institute of Technology discussed the much-mooted subject of rationalized load factors. The desirability of pricing loads on fundamentals in load factor specifications and of establishing some degree of consistency among the requirements for the various parts of an airplane was a major item of interest in the discussion before the Department of Commerce and the representatives of the industry in Washington in July (see *Airways* for September, page 537), but no attempt to go into detail was made at that time. Professor Newell said that he was in Cleveland with suggestions for definite formulas, for putting the load factors in terms of those elements of flight which should be considered actually to control the magnitude of the largest loads falling on the structure.

He made his suggestions that it is quite impossible to build an airplane to stand the largest loads that can be imposed by the most violent imaginable use of the controls, unless at the same time the machine be made almost impracticable in maneuverability. There is plenty of evidence to show that a single plane can be pulled up from dive to the angle of maximum lift with such rapidity that the load factor will be only about ten per cent above the theoretical maximum figure. For typical piston planes pulled up sharply from a vertical dive that would correspond to a load on the wings of about 25 times the weight of the machine in the steady load, an absolute maximum is imposed by the physical limitations of the pilot. So far as there is evidence on the point it appears that the lower limit cannot stand a load factor much, if any, in excess of six without danger

of serious injury. Allowing a true factor of safety of one and a half, piston planes need not then be designed for a load history of more than about seven in any event.

It is clear, the author pointed out, that wing loads tend to increase with increasing maximum speed of flight and also with decreasing wing loading, whatever the loads be due to maneuvering or those due to flying through bumps.

When it is immateriality that in order consideration the power of the control is of course a factor, determining whether or not the pilot is able or likely to pull out at a given very early stage. As a measure of control Professor Newell accepts the landing speed constant, or ratio of the horizontal tail surface area to the wing area multiplied by the ratio of the distance between the center of gravity of the airplane and the tail surfaces to the wing chord, or

$$K = \frac{S_1}{S_2} \times \frac{l_1}{l_2}$$

Professor Newell then recommended a value of 4.5 K for low load factor in commercial planes, or 4.5 K/3 in military types, it being the control constant and K the speed ratio ratio, and showed that in most cases the formula would put load factors within ten per cent of those now required by the Department of Commerce. Of course that is in itself, with an underlying fact, is not a very strong argument in favor of any formula, for the search for a new method of specifying load factors is naturally based on the assumption that the present method is not as good as it might be and it is not in all cases correct. It is also reached and the present method throughout in their minds there would be no reason for making a change at all.

The author next turned to the much more troublesome question of near speed loads, as determined by the so-called "low angle load factor." He discussed the fact that if the machine be pulled up sharply from a dive at terminal speed, the load on the rear spars will be less than on the wing spar, and steadily with increasing angle of attack until the angle of maximum lift is reached. The specifying of near speed loads in terms of a constant g from angle of attack can be justified only on the assumption that in otherwise out from a dive the first 4 or 5 g's at change of angle of attack will be accomplished without appreciable loss of speed, while thereafter the g -factor will rather be so gradual that the load factor will be pulled up from zero rather than increase in rear spar loading. The assumption seems reasonable, although it is not there is less experimental data to support it as to continuous g .

The paper included no direct formula

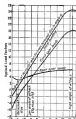


FIGURE 2. Lift factor curves for a low airplane. The plane is shown in the position of the wing in the figure.

for specifying the low-angle load factor, but it contained a suggestion that the instantaneous pull-up from the diving attitude to the angle of attack at which the lift coefficient has one-quarter of its maximum value should be assumed in all cases. This would give the formula

$$\text{Load factor} = 1 \left(\frac{V_1^2}{V_{\text{min}}^2} \right)^2$$

where V_1 is highest speed of first that would ever have to be achieved with the use of present engines, with which low vertical dives are made, the terminal speed may rise to six times the maximum speed, giving a low angle load factor of nine. Professor Newell suggested that for commercial types the maximum permissible diving speed should be determined and based on the maximum load, with the warning that it must never be exceeded, the low-angle curve then being calculated on that basis. These are of course some commercial objections to such a position, for experience shows that any increase in a machine's structural limitations or any suggestion of the desirability of any measure of restriction in its use, immediately arouse in the pilot's mind a suspicion that the machine must be structurally weak in order to require such warnings.

Tail loads

The paper contained no comment on load specifications for fuselages, landing gear, cinders, or other, but experience and common sense did come in for criticism. There are two recognized methods for determining the tail load when flying from a dive. That of A. V. Kober of the R.A.C.A. reads:

$$\text{Load} = \left[0.034 \left(\frac{S_1}{S_2} \right)^{1/2} \right] 5.7 V^2$$

while another developed by Lieutenant February and Gilman, while studying under the supervision of Professor Newell, reads:

$$\text{Load} = 7 + 0.030 \frac{S_1}{S_2} V^2$$

where S_1 is the area of the elevator and S_2 that of the rest of the aircraft tail surface. The wing chord is the distance from the center of gravity to the center of pressure of the tail surfaces. V is the diving speed from which the sudden pull-up is made, and l_1 the distance from the tail to the equilibrium is a steady dive at speed V . The February and Gilman formula is based on the assumption by the tail in a straight dive, with the pulling load upward on the elevator supported on it, the elevator being considered as a plane with a pressure of 25 degrees for determining the additional increment of load.

In practically all cases these two formulas show great agreement, and a typical piston machine gives plus a tail load load about equal to the tail

load in a terminal dive at limiting velocity without allowance for pull-up, the diving attitude which British designers are required to meet.

Inverted flight conditions were apparently eliminated from consideration in the paper for lack of experimental data. Professor Newell urged the desirability of more study on the effects of suddenly pushing the stick forward and so both in inside loops and similar maneuvers. He thought it very likely that two inverted conditions would adequately be required, corresponding to the low-angle and high-angle conditions for inverted flight.

Catapulting on a large scale



Collecting a 7-ton biplane into the air with a 100-ft. cable. The plane is seen in flight above the deck of the aircraft carrier. The left view shows the down and up position of the catapult.

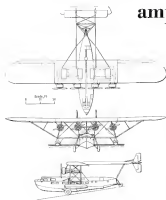
THE practice of catapulting airplanes off most warships has been the subject of much discussion in the past. The Royal Air Force conducted an experiment in which a 7-ton biplane was catapulted from the deck of a ship. The catapult was a cable system attached to a drum which was rotated by a motor. The plane was launched at a speed of 60 mph in 100 ft. in less than 10 seconds. The plane was then recovered by a crane. The experiment was a success and showed that catapulting was a practical method of launching aircraft from ships.

planes are run up to flying speed and the compressed air forced into the catapult. In the case of the plane shown a speed of 60 mph in 100 ft. in less than 10 seconds. The plane was then recovered by a crane. The experiment was a success and showed that catapulting was a practical method of launching aircraft from ships.

The Sikorsky S-40

amphibion

A new giant for the
Pan-American
air fleet



nons have hitherto been considered impractical. It is only natural, therefore, that the various networks, having worked out their individual problems, should turn in the greater care of inter-connection, and the establishment of world-wide trade routes of the air.

One of the most important aid in the same line, one of the most difficult interconnecting routes is that between Europe and South America. With a new breed of the establishment of mail and passenger schedules over the Atlantic, Pan American Airways has been making an extensive study of means of connecting with the Imperial Airways via Bermuda and the Azores. They have, during the last four years, equipped themselves with experience and equipment for the relatively long over-water stages between the United States, and Central and South America. Their equipment has been limited, however, to airplanes of a gross weight of 7 or 8 tons, and a cruising range of about 200 miles, but sufficient has been available to connect with the long-range flying boats developed by England for the India-Australia, or the South African routes. Although present schedules do not yet afford sufficient passenger volume to justify the purchase of airplanes of a comparable size, yet it was considered that a replacement for long over-water flights with heavy loads was sufficiently imminent to warrant the authorization of two experimental amphibious flying boats of a size far beyond any which had hitherto been designed and built in the United States.

Such there has been no gradual development from smaller units to ones of the size contemplated it was necessary to bridge in one jump the gap between the design of an amphibious 5-ton gross weight, and one of 27 tons. An order was placed with the Sikorsky Aircraft Corporation, however, for the first two machines of this displacement, the first of which has been successfully test flown, and is now ready for delivery. Neither of the boats is intended for trans-Atlantic service, but they will be constructed on the 1300-mile over-water route between North and South America where their performance will be clearly stated with a view toward the development of equipment for the longer trans-oceanic flights. With passengers and freight shown under load, even in attacking the problem of the housing and handling of airplanes on such a scale at the present time.



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General specifications

The Sikorsky S-40 is a high-wing amphibious airplane of the same general type as the well known S-41 which is characterized by the carrying of the tail section on a pair of cantilevered floats. The Pratt and Whitney four-cylinder Horner engines of 275 hp each mounted as tractor on nacelles hang below and forward of the leading edge of the wing beneath the power. A single van-bottom hull containing all accommodations for passengers and crew is supported below the wing, and a pair of outboard wing floats hold stabilize on

the water. Fully retracting landing wheels and tail wheel have been provided, but the machine may be flown as a flying boat with considerable improvement in performance, and with some 1000 lb of additional payload available. The present arrangement of the machine and its overall dimensions are shown in the three-view drawing. Figure 1, as an amphibian, the empty weight is 11,500 lb, which, with a designed gross loading of 56,000 lb, allows some 12,500 lb for useful load. On the basis of the gross weight, the wing loading is then 132.2 lb per sq ft, and the power loading 14.8 lb per hp. With accommodations for 24 passengers the S-40 is expected to have a range of approximately 300 miles and with the maximum load of 40 passengers, a useful range of about 300 miles. On test flights with full load the machine has attained a high speed of better than 130 m.p.h., cruising at 110 to 115 m.p.h. and loads at approximately 65 m.p.h. A rate of climb of 712 ft per min. was obtained, and an absolute ceiling of 14,800 ft is predicted. A cruising floor Pratt and Whitney four-cylinder Horner engines of 275 hp each mounted as tractor on nacelles hang below and forward of the leading edge of the wing beneath the power. A single van-bottom hull containing all accommodations for passengers and crew is supported below the wing, and a pair of outboard wing floats hold stabilize on

by Sept. 1, 1941 it is expected that the first machine will be ready for delivery to the Pan American Airways some time before Oct. 1.

Ball and wing floats

In the preliminary work on ball design for the S-40 (which dates back to April, 1939) twelve models of varying shape were experimentally tested for aerodynamic and water-resistance characteristics. Part of this work was carried out in the Washington Navy Yard towing basin, but the greater part was done in the Hunscombe River by towing models alongside a motor boat (Type 205, Admiralty, May, 1939). In the reported design the van bottom has two rings, one located approximately at the mid-point of the 28 ft ball, and the other ring about 18 ft from the stern post. The portion of the bottom just ahead of the after ring has been given a decided bow-down, and the last five six of the rear ring steps sharply upward to give adequate water clearance when the machine is running on the ship.

The side plating is vertical, and perfectly flat from chine to gunwale. The chine over the stern is well rounded from gunwale to keel, and is gradually flared along the top longitudinal chine for the entire part of its length. It is rounded down over the cockpit, and tapered down to the stern post in a sharp V-shape at the rear view hull-head. A sharp action forward of the cockpit is about half below the main deck level to permit adequate plan view forward. The gross submerged

displacement of the hull is 100 tons.

The construction is all metal, non-performing rounded streamline shapes and Alclad sheets, assembled with steel rivets. The principal structural member of the hull is a deep girder-type hull to which the built-up members (frames) are attached at relatively short intervals. Light members spaced on short centers strengthen the transverse frames. Six glass-type bulkheads, each equipped with a watertight door, divide the hull into seven watertight compartments, any two of which may be completely flooded out the pump room above. At the point where the stern wing trailing structure is fast to the hull a shock-type bulkhead has been substituted. The framing in this point is made up of heavy round steel tubing assembled by means of special flanges. It will be surrounded by a light, non-watertight partition. The construction is clothed in one of the non-performing characteristics. Garages are deep streamline channels, mounted with flanges upward and with side shells cut for lightness. The hull plating is riveted directly to the transverse framing, and is reinforced longitudinally by angle members. All seams are sealed with fabric and marine glue during assembly.

Three persons stand behind the gun turret in the interior of the hull. The after hatch is designed to be the main entrance, and is located so that passengers may embark either from a landing, or from the ground. The cover swings outward and downward to form a short series of steps for the convenience of the passengers. The other two

hatchways are for emergency use, for the loading of unusually bulky items. Five manhole ports, and eight rectangular windows are provided on each side. The latter are large enough for emergency exit purposes. They are closed with non-flammable glass plates in frames so arranged that they can be swung outward from the bottom to afford exit lanes. Extra reinforcement has been included around the window openings in the hull plating to reduce the possibility of windows being wringed shut in the event of a crash.

Incorporated in the hull structure are several steering and novel features. A rectangular channel houses the engine control cables and pulleys in isolation from the engine and cables outside the hull proper. This arrangement permits easy access to the controls without disturbing the hull structure or cable bearings. A short distance aft of the end of the channel and just forward of the after hatchway, an engine mounting ring is built into the cable rail structure. Normally a circular cover prevents the continuity of the deck line, but in case where a stout engine is required at some

outlying repair station, the cover may be removed, and the engine lashed down to the mounting ring in the deck. This construction allows a very convenient way of transferring such spare equipment to non-performing without interfering with the hull, or without the need for complicated handling arrangements to lower an engine into the hull.

Removable landing lights are set in the side plating of the hull ahead of the cockpit, and a small hatch in the forward decking gives access to the engine which can be downed up into the lower part conveniently by one man using a small crane. A bumper at the usual Skyrocket type is set in the bow.

The wing bracing is of the same general construction as the main hull, being made up of streamline framing members covered with Alclad sheet. They are also of the box-beam type, with a single spar. The most interesting feature is the incorporation of a 240-pai gasoline tank in each. The tanks have the same width and cross-section as the deck of the float, and the control portion of the docks between the points of attachment of the supporting strut straps is recessed to form a sort of socket into which the tanks fit to complete the wing box laterally, and fore and aft. Since the tanks are not tapered with the float there is little likelihood of loss or contamination of gasoline due to damage when taking off or landing. The floats are carried on streamlined horizontal struts which extend outward from the hull, and form a portion of the wing truss system. A typical outward fitting is shown in an accompanying sketch. The interstrut configuration of such float is 7,250 lb., and after effective righting factor is 4.2 with full load.

Wings and tail surfaces

The monoplane wing of the S-48 is made up of a center-section with a span of 64 ft., tapered directly to the tail,



The after hatch with cover open. Note also the deck of the hull plating and the mounting ring fitted in the deck by the coupling of a spare engine.

carrying the tail outriggers and the four engine nacelles, and two tip sections, each 24 ft. 11 in. long, making the overall wing span 114 ft. The innering is built up of extruded duralumin sections, and forward dural sheet with a certain amount of steel plates at various fittings. The entire structure is fabric covered with the exception of a metal skinned portion of the center section over the engine nacelles.

The wings are of the conventional two-spar type. The inner and outer beams are of a deep girder type construction made up of extruded channel flanges and formed sheet channels, connected by gusset plates. Drag struts of round construction connect the inner beams at 40-degree intervals. The ribs are built up of extruded tee and channel sections, the former being used for tip struts, and the latter for web bracing. Inner elements of the balanced type are mounted in the tip sections.

A brace carrying the entire wing is supported by cantilevered brackets extending forward from the rear spar at each drag strut. All control bracing, as well as all other bracing and control pulleys on board, are built inboard.

The aileron control is of the cable and pulley type, the cables from each aileron running in toward the center section and hooked to the outer spar. When the outer section is attached to a pair of yokes which in turn are connected by links to a large cleave as shown in an accompanying sketch. Control wires from the cockpit come up through the nose of the hull. Rotational movement of the cleave produces a rotating motion of the yokes, which is transmitted by the cables to the ailerons.

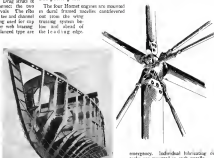
Tail surfaces are of the standard Skyrocket type, including fins and rudders of special slatted section to correct for unbalanced air stream effects due to stoppage of one or more engines, and use of the same general construction as the wing, being fabric-covered over dural framing.

Control surfaces are supported on both elevators and rudders, and sufficient adjustment is obtained by rotating the entire tail assembly, including the fins and rudders, about point of attachment to the outriggers.

The latter differ markedly in construction from the usual Skyrocket fabric-covered fin type. They consist of duralumin box girders built up with plate webs and are for flapless. Numerous longitudinal and transverse angles are riveted to the inside surfaces of the webs close to wing surfaces. In order to make this type of construction feasible, each beam was made up in symmetrical halves and assembled by placing them together and connecting up the outside flanges at the top and bottom flange are built in.

The power plant

The four Pratt & Whitney are mounted in dural formed nacelles cantilevered out from the wing trailing system below and ahead of the leading edge.



Four view before flying showing hull form, and construction of the main transverse framing and bulkheads. Right: A typical wing, lower rib shown, and wing float bracing joint with flange recess.

Each nacelle is well streamlined and is engine surrounded by a Skyrocket type ring cowling. Two-Madell Hamilton Stand, are popovers, 18 ft. 6 in. in diameter are driven directly by each engine.

The tail sample of approximately 1300 psi is formed in one tank, four of which of 147 psi each, are located within the wing immediately above each engine nacelle, and two others, each of 240-pai capacity, are located in the wing flange as described previously. Fuel tanks are arranged that any engine or group of engines may be fed from any one or any group of tanks. All valves

and pumps are located outside the hull and are operated by remote control. It is thus possible for gasoline fumes to enter the cabin and cause discomfort to the passengers. This feature makes possible also the permitting of smoking in the cabin without danger of fire. Electrically driven pumps, located at the fuel tanks, raise the fuel to the wing tanks as required. A hand pump is provided for use in case of failure of the electric pumps. The fuel tanks are protected with self-extinguishing damp rakes, so that their fuel load may be jettisoned safely, or in part, in case of

emergency. Individual fuelizing of tanks are mounted in each nacelle.

All engine controls pass from the pilot's cockpit through the cockpit in the roof of the hull from which they lead upward to the center section of the wing through a streamline housing. From the center section the controls are distributed to the various engines by double cables carried along the front edge of the forward wing spar. Every each point of attachment of the engine the controls are directed downward into the nacelle by means of an ingenious arrangement of pulleys and drive segments anchored to the front cowling. It is interesting to note the use of heat-treated dural cushions for various parts of the control system. All such cushions and all fittings are fitted with gaskets. Loads of any sort in all parts of the plane were relieved by X-ray examination before being tested.

Amphibious gear

The amphibious gear supplied with the S-48 is of the usual Skyrocket type, and is suitable largely on account of its



Cabin interior showing streamlined bulkhead construction.

relatively great size. The two main wheels are of cast iron, constructed and carry hydraulic-operated brakes similar in design to those used on the S-38. Due to the high pressure required to operate the brakes of this size, a servo system has been introduced whereby the hydraulic brake system is actuated by compressed air which, in turn, is controlled by valves at the brake pedals. The elements of the linking system are shown in one of the accompanying figures.



Schematic diagram of brake system

all regions while the plane is in flight.

One of the most important problems from the standpoint of passenger comfort is reduction of noise, for which purpose the entire cabin is being lined with a thick blanket of insulating material between the outer shell and the inner gas-tighting. Observers are located in the plane during test and flight; reports that even without any insulation in the cabin with the external noise was greatly below the level of that in the average transport airplane and that

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very little vibration was experienced. Because it is believed that a certain amount of vibration enters a hull from outside propulsion, special reinforcement has been provided in the case of the hull in the section which lies in and adjacent to the plane of rotation of the propellers. Other factors which are planned in lesser degree include, close in the cabin are the relatively remote location of the power plants, and the vibration-absorbing capacity of the relatively large mass of the structure. A well-sprung pulley, including elastic stages and a suspension, has been installed, and separate ventilators for men and women are provided. It is not the intention to operate the plane crowded to absolute capacity in passenger, but rather to limit the accommodations to the point where each passenger has adequate room to sit or more about in the greatest possible comfort.

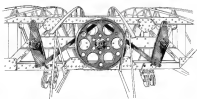
Particular attention has been given to the selection and arrangement of the furnishings and fittings of the cabin. Every scrap of material which has been used inside the hull has been made to conform to the highest standards of safety.

Decorations

The problem of interior decoration was placed in the hands of Mrs. Guy Mayhew, who has made the furnishing of aircraft her hobby. She has made a study of suitable colors and interiors and has made a large number of expert, exact measurements to determine the most appropriate color scheme for the S-40. She was allowed a maximum of 1,000 lb. to include wall coverings, carpets, seats, cushions, lighting and wire glass fixtures, and other equipment. By a combination of light natural finished wood paneling, oil finished silver lacquer, blue carpets, and orange lacquer, a most effective and spacious interior has been achieved which suggests comfort and convenience without intrusion of mechanical aspects of flight.



The throttle control in the cockpit



Differential control for altitude on rear part of the engine section. Valves from engine are closed through valves, and valve action valve action in rear section.



The largest dirigible

A brief but comprehensive description of the U.S.S. Akron's design and construction

	Estimate	Actual	Notes
Overall length (ft.)	425.00	425.00	
Overall width (ft.)	100.00	100.00	
Maximum diameter (ft.)	100.00	100.00	
Height (ft.)	100.00	100.00	
Weight (lb.)	100,000	100,000	
Number of engines	10	10	
Total horsepower	1,000	1,000	
Maximum speed (m.p.h.)	10	10	
Range (miles)	1,000	1,000	

* At 10 knots cruising speed, without refueling.

LESS than one year after the greatest dirigible in history—on the basis of the immense size—was built, the U.S.S. Akron is now in the hands of the U.S. Navy. The Akron is a new project in the class of the ZEPPELIN, better known as the U.S.S. Akron. This ship is not only the largest of the type, but because it is larger than any yet launched, represents a new step in the construction and operation of the dirigible. It is the first actual design of the Goodyear-Zeppelin Corporation and is now in the hands of the U.S. Navy. The general theory of the design is conventional. That is, there is a shell or hull made up of numerous frames and longitudinal supports by a series of balloons. In structure and in detail arrangement, however, there are notable innovations.

The first feature of the design is the placing of the engine within the hull rather than in gondolas or nacelles where the other dirigibles and

provide heated working room for mechanics. This is the first actual design of an old Zeppelin idea. There are eight 50-hp. engines compartments within the dirigible, each equipped with the main mechanism and a specially built blower to provide fresh air for the engine units, and for ventilation of the structure and the crew. There is also a wheel by which the angle of the propeller shaft may be controlled.

The solid engine arrangement with the use of a 30-hp. propeller shaft in gondolas which has 600 tons thrust the vessel to a speed of 10 knots.

directly to the propeller. Still another innovation has been introduced, but in working the gas through a filter to remove any of 50 degrees from the horizontal. This permits obtaining from the engines a total upward thrust of about 6,000 lb. or a downward thrust of about 10,000 lb. The engines are reversible.

The transmission system gives a speed reduction of 7 to 4 to a constant speed of 1,000 r.p.m. because a propeller speed of 500 r.p.m. The two-bladed propellers are especially designed to be effective in the forward or backward motion of the dirigible.

The engines themselves are an innovation. They are Mayhew models. They are 30-hp. V-12.

Flying Equipment

SERIES B-1 HORNEY ENGINE

ANEW radial air cooled airplane engine has recently been completed and tested by the Pratt & Whitney Aircraft Company, of East Hartford, Conn. Although based primarily on the Series B Hornet engine which was first introduced in 1935, the new engine, the B-2, incorporates several modifications, notably a 1:2 propeller drive reduction, higher compression ratio, and improvements in the supercharger drive and oil temperature regulating system.

The new reduction gear mounts primarily at a drive gear splined to the crankshaft which drives six pistons on a crank in a cage bearing on the crankshaft and automatically regulating links so that each leaves its designed share of the load. The floating gear cage engenders the movement tendency of the crankshaft and provides for equal strain on the driving and the driven gears under all conditions of operation. The piston drive is a lead gear which is part of the propeller shaft. Bulk pinion, pinion shaft and pinion gear are mounted on roller bearings and the entire reduction gear housing may be removed with the propeller shaft for inspection.

In the case of the 322-hp. Hornet during the most modern department test stand practice is the use of a special cage bolted to the blower section supporting the front of both the intermediate and the propeller shafts. In the standard drive the intermediate shaft has no main bearing and the propeller shaft is supported on a ball seat in the blower section, but this requirement for higher blower speed in the new engine necessitated an increase in bearing size.

The floating gear of the drive incorporates two sets of teeth, one of which engages the crank on the one side of the crankshaft and the other drives the intermediate blower gear. The floating gear is mounted on a roller bearing and is supported on the outside of the main gear, which in turn supports the propeller shaft and the intermediate drive shaft.

Special pistons, fitted in the B-1 Hornet give a compression ratio of 6:1 instead of the standard 5:1. With its speed in oil temperature control, a com-



Piston cap of 1:2 reduction gear oil supercharger piston.

pression hot-spot and oil regulator in use which provides access not only to heating the pistons in cold weather but also for cooling the oil by passing it through a valve which is placed in the cylinder cooling system between the cylinder and the supercharger. An internally-operated bypass valve provides a positive means of unloading the oil supercharger within any defined limit.

According to the manufacturer, the new engine is rated at 325 hp. at 2,600 rpm. at 6,000 ft. altitude. The weight, including carburetor, running gear, compression hot-spot and oil regulator, air scoop, generator drive and propeller shaft attachment parts is given at 915 lb. On this basis the engine weighs 175

lb. per hp. at 5,000 ft. and 148 lb. per hp. at sea level.

GREAT LAKES SPORT TRAINERS

THREE two 1500 models of the well-known Great Lakes Sport Trainer follow very closely the line laid down for these machines in previous years. The chief improvement in external appearance is around the nose where careful attention to engine cooling and landing has resulted in better lines. The new models are offered with optional power plants, either an upright four-cylinder, Chrysler or the second four-cylinder Corvair design. The exhaust stacks provided with both models are fixed for cooling. Both machines are open-cockpit lightplanes designed for tandem seating of two persons. The upper wings are rigid with considerable overhang and an dihedral, whereas the lower wings are rigid without overhang and with several degrees of dihedral, resulting in considerably more stagger at the outer section of the wing tip. The material and the type of construction



The Great Lakes Sport Trainer with Corvair engine design.

are conventional throughout. The general characteristics of the two machines are similar and are given by the manufacturer as follows:

Seach speed	120 k. t. h.
Model	1500 k. t. h.
Engine	1500 k. t. h.
Wing area	1,000 sq. ft.
Wing span	30 ft.
Wing chord	10 ft.
Wing weight	1,000 lb.

NICHOLSON JUNIOR

A NEW light monoplane has been designed and recently test flown by Mr. J. L. Nicholson, Jr., of Timpanish, N. Y. The machine is a high wing, single-place, semi-cantilever monoplane powered by a 40-hp. Bashly engine. It is intended that later models will be fixed with seats for two persons, as preliminary test flights have indicated that an additional 170-hp. load may be carried safely in the single-seat experimental machine.

Material and construction are conventional. The fuselage is built up of welded chrome molybdenum steel tubing as are the landing gear and tail surfaces. Wings consist of laminated spruce spars, Duralumin ribs, fabric covered, and aluminum are built on spruce spars with aluminum ribs, also fabric covered. The undercarriage is fixed with shock absorber struts and Good-year Aerofoils. The general characteristics of the new two place model, as given by the designer are as follows:

Seach speed	120 k. t. h.
Model	1500 k. t. h.
Engine	1500 k. t. h.
Wing area	1,000 sq. ft.
Wing span	30 ft.
Wing chord	10 ft.
Wing weight	1,000 lb.
Wing loading	100 lb. sq. ft.
Power loading	150 lb. sq. ft.

BELLANCA SKYROCKET DELUXE

A NEW model of the well-known Skyrocket has recently been announced by the Bellanca Aircraft Corporation, of New Canaan, Conn. The latest design is similar to that of the previous models first introduced in 1930, but increased

speed has been obtained by the addition of extra features and refinement of details.

The machine is powered with a Pratt & Whitney Wasp series 5C engine with a 14:1 compression ratio and a 30:1 blowdown ratio, reported to develop 480 hp. at 2,100 rpm at an altitude of 5,000 ft. In addition to the increase in horsepower, the machine has been fixed with a Segt-type cowling, and the same cowling has been improved around the engine to give better stream-line effects. Landing wheels, including the tail wheel, have been completely redesigned in stream-line form, and all exposed fittings have been provided with generous fairings. A special tail valved tank has been fitted to all engine surfaces to reduce air friction. The general characteristics of the machine, with the ex-

ception of the change in horsepower ratings, are the same as those of the earlier Skyrocket model.

APPROVED TYPE CERTIFICATES

DURING the period Aug. 15 to Sept. 15 the Aeronautics Branch of the Department of Commerce issued the following Group I approved type certificates: 440, American Pilger 150A, (Aircraft 150A hp 1, 440, Cyclone C-1 (Continental A-25 25 hp 1); 441, Pinner Antares PCA-3 (Wasp Junior 300 hp 1, 442, Aeromarine, Aeromarine C1 (Aeromarine E113 25 hp 1); 443, Aeromarine, Aeromarine C2 (Aeromarine E113).



AVIATION, by Louis C. J. Maguire, U. S. N., McGraw-Hill Book Company, New York, 1951, 136 pages, \$7.50.

BOOKS on meteorology often err. From the point of view of the layman, when he goes into involved discussions of the physics of atmospheric phenomena or in simplifying the subject so much that they lead down to a few general rules for forecasting and reading weather maps, with almost no explanation of the underlying facts. Leonard Maguire, a graduate of Princeton University's course in meteorological technology at the Massachusetts Institute of Technology and now associate for the Bureau of Aeronautics, has avoided both extremes. His book is written in terms of observed phenomena rather than of partial differential equations, but the phenomena are discussed with reference to their cause. Only one chapter, and that a very short one, is

given over to weather forecasting, while weather, somewhat longer, discusses the construction of weather maps. Most of the author's attention was given to the relationship between pressure, temperature, wind, fog, and precipitation, and such matters as the tendency of the winds to blow in spirals. In the Eastern Hemisphere's volume cannot be said to stand out as the one good book on the subject, but there have been other good ones from various points of view, at least it is one of the best.

TECHNOLOGY AND EXPERIMENTAL (Practical Book of Aeronautics) 1951, compiled by Dr. Werner von Langsdorff, 131 vol., military, commercial, and general aviation, 1,000 pages, \$1.50. L. J. Lohmann, New York, 1951; 128 pages total, 5 marks each.

THREE well-known annual reports for 1951 in this volume, each a comprehensive and well-organized survey of a specific class of aviation. They are of considerable value, being quite different in this respect from the classic "The World's Aircraft."

The present annual differs from others, also, in being in three languages so that lack of knowledge of German need deter no one from using it. Each volume includes photographs and a standardized table of specifications of some 400 airplanes, together with a general specification index. The calculations are



The Skyrocket Deluxe

complete and well prepared, and an interesting and commendable feature in the inclusion of the name of the engineer responsible for the design on each case. The American section was prepared with the special assistance of A. A. Gossert, lately chief engineer of the Fisher Company, and Capt. A. M. Johnson, the Director of Naval Ordnance, and is very satisfactory except for a few minor omissions of names, and, especially in British. The three volumes are strongly recommended as a useful addition to every aviation library, and to every engineer's personal reference collection.

ELASTIC ENERGY THEORY, by J. A. Faa Du Breuil. John Wiley & Sons, New York, 1931, 342 pages, 14.50.

THIS treat on the strength of materials goes more deeply, and much further, into advanced methods than is conventional. It concerns itself especially with statically indeterminate structures, redundant frames, systems of beams, loaded frames, and methods dependent on analysis of deflections as means of determining such as the method of least work. A number of examples are used, and three of them are of specifically aeronautical interest, the latter being devoted to the analysis of a typical airplane fuselage, how to a consideration of the general stressing process in an isolated wing (monocoque), quite superficially considered and without reference to secondary influences), and one in a consideration of a fuselage loaded with airplane stress like the war-time Fokker.

REPORT ON THE EXPERIMENTAL RESULTS OF THE ORDONANCE SERVICE PLANS WITH THE AIR OF FRANCE TAKEN FROM THE AIR, W. H. Perry. O.R.C., London, England, 1930 and 1931.

THESE documents present principles and conclusions of two periods of experimental consideration of aerial mapping with the specially created ground methods followed by the Ordnance Service, work which corresponds completely to the (topographical) mapping of the United States Geological Survey. The first report indicates that aerial surveys over flat and west more accurate than profiling methods over uneven or flat country. The second report and third book, the second, based on longer experiments with the air method is less favorable. It is considered somewhat only in cloudy light some areas and even less data are complete. It is also found that all work had to be checked as the ground in order to maintain standards of the department. It should be emphasized that these maps show every detail even down to wire fences and partitions between houses, extremely detail beyond the range of aerial photography.

EXPERIMENTAL AERODYNAMICS, by T. G. Whitcomb. Oxford University Press, New York, 1930, 250 pages.

THIS book is written to appeal to the student of aeronautical engineering, who is primarily concerned in obtaining a working knowledge of the aerodynamics of airplane design and who would be not sufficiently grounded in mathematics to be able to follow the applications of hydrodynamics to aerial flight.

This makes it necessary for the reader to accept without analytical proof, several equations derived elsewhere. However, since a great deal of experimental evidence is given to check these relations, there is no great loss of rigor by the student of mathematical derivation that are beyond the comprehension of the average engineering student.

The contents, briefly, are as follows: airplane nomenclature, the velocity atmosphere, the wind in two dimensions, flow, airfoil combinations, gas-dynamic performance estimation; equilibrium, stability, control, maneuverability; the propeller, and scale effect. An appendix giving the collection of the angle of attack, and the constant coefficient in area lift is also included.

The book is well written and gives much in value because of the larger number of illustrative problems which are worked out in the text. Moreover, at the end of most of the chapters there are problems to assist the student in applying the subject matter of the chapter. It refers to a complete list of references. Current and material data are reproduced without any mention of their source. An index of references would aid the student in reference reading and would also give credit where credit is due.

The contents of this matter is also somewhat erratic. The navigation deals with problems the student would be likely to encounter until he had read Chapters II and VI, the subject of the coefficient of the airplane is partly dealt with in Chapter II, and partly in Chapter IX and Chapter XII, an index effect, should follow soon after the introduction, since experimental results are related to experiment. The first report indicates that it is not clear why the subject matter of Chapters II and V should not be combined, since both deal with airfoils and planes, the position of the airfoil characteristics should be given in terms of two dimensional flow—P. W. G. Gossert, [Aeronautical Institute of Technology].

COCKNEY FLYING, by Maj. Oliver Street, Richard R. Smith, Inc., New York City, 1931.

WRITTEN specially for the amateur pilot, this book gives a lot of the essentials in the art of flying.

ing his way across country under most conditions he is liable to encounter. Since aerodynamic flying problems have been more than solved, even the earliest days of man's flight being able to serve as a Royal Flying Corps pilot across the Channel, the subject obviously is handled thoroughly and thoroughly. In fact, a perhaps goes into more detail on certain points than is necessary for the student.

It is emphasized that a book on this subject has been prepared chiefly for the new pilot, and is not intended as a review of the growing importance of this branch of aviation. Any pilot flying an aircraft cross-country must, and especially the non-pilot, have a device which will indicate him from the volume.

ENGINEERING, by T. G. Whitcomb and Maj. Oliver Street, Richard R. Smith, Inc., New York, 1931, 352 pages, 12.50.

THIS second edition of Mr. Street's book has been brought up to date, and shows the results of his constant study of the material and developments presented by the Aeronautical Model League of America. Although the book is described as devoted to model airplanes, it attempts to qualify as a general manual for introductory sections on conventional flight and on airplanes and engines. Both theory and practice of model construction are well covered, and descriptions of a dozen typical model airplanes are coupled with detailed instructions for building them. Suggestions for acquiring model data and conducting contests and a list of references on flying models are included. It is in the form of a very readable, and will be very useful to serious people who, however skilled or not to the local Boy Scout troop, are in actual contact with the spirit of the aviation game in their own knowledge of how models are made to remain aloft and fly. It is a book with the power of a trained student of aviation.

STREET FLYING, by Capt. Richard Street, Gordon & Breach, London, 1931, 112 pages.

THIS book is another treatise on the various steps in learning to fly, and is written by an expert of the type, for which the book is needed, and deals with problems in regard to all other experiences flying of the standard type. It is rather one of place in modern aviation, and the emphasis is upon the student. The contents of chapters, flying, and the student's operations are presented in addition to those for land flying. As a matter of fact, street flying is covered only a comparatively small part of the text and the title is rather misleading. It is not a particularly good introduction to either the real or the apparent subject.

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Design Novelties

Inspection doors for tail controls

AMERICAN and effective arrangements for the inspection and adjustment of tail surface controls is incorporated in the Massachusetts composite shown in the German light plane case for 1931. The after-portion of the fuselage door just forward of the fin is connected with a pair of control cables, each hinged along the upper longitudinal, and meeting in the same line of the fuselage. The connecting a straight metal pin the two sections may be drawn back over the top of the fuselage, thus giving complete access to the air-portion of the barage.

Jury strut on the Avian

BEFORE folding back the wings of the Avian Aero biplane for storage it is necessary to provide some means of preserving the alignment of the upper and the lower wings when the front spar hinges are detached from the rear section and the fuselage. For this purpose, a jury strut is provided which is normally extended folded back along the lower surface of the upper wing, after the spar hinges are detached from the rear section. It is made up of two pieces of steel tubing with a telescoping joint. Before extending the hinges of the wings it is swung downward and extended so that its lower end, which is provided, engages in a shallow socket under the front skirt of the walkway on the lower wing. Internal threads are provided at the telescoping joint so that the strut can be set in different compression between the upper and lower wing panel to preserve the alignment of the structure.



Member's design inspection doors.

A single strut landing gear



The rear section of the Avian biplane.

serve the alignment of the structure after the front hinge joint has been removed.

Improved streamlining for the Condor

THE top speed of the new Condor recently delivered to Eastern Air Transport, Inc. has been improved by 12 to 15 m.p.h. as compared with the older models by redesigning the fuselage through the tail. Condor has adopted the transport use from a design intended as a military bomber, the engine nacelles were last slightly modified, although each originally had provision for a machine gunner's cockpit in the tail.

The new nacelles are somewhat larger, but not quite as deep as their prototypes, and have been designed without regard for military purposes. They have been somewhat streamlined from nose to tail, and provided with gunnery holes where the observation area of the fuselage is long. The original indicator location has been retained.

TO CUT down the resistance of the undercarriage on the high speed V-type transport, recently developed for the United States Army, the Lockheed Aircraft Corporation has designed an interesting single strut landing gear. Each wheel is carried on a single streamlined strut hinged to the fuselage with four sets of streamlined struts. The rear strut, but no shock absorber, The wheel, which carries a standard shock and a large low-pressure tire, is mounted on an axle mounted on a steel bogey. The latter is of the shape of a square block and incorporates a shock absorber unit parallel with, and



on either side of the shock absorber unit, two short rubber members are mounted so that they are hinged to the wheel block and also through lugs in a hinging attached to the end of the rigid landing gear arm. When assembled, the rubber members act as a cushion between the wheel block and the fuselage, but permit the up and down movement of the wheel with respect to the strut. The vertical movement is controlled and limited by attaching the free end of the shock absorber unit to the arm flange. The entire wheel assembly has, therefore, freedom to move up and down against the resistance of the shock absorber unit. The energy to be absorbed in landing is distributed between the low pressure tire and the shock absorber.

Aircraft at Work



Landing event and photos in broad beam

Dusting used effectively against timber pest

A second airplane dusting operation recently completed in the state of Washington, has brought excellent results in the struggle to eradicate the "hemlock looper," a destructive insect pest. Northwest Air Service, Inc., Seattle, was contracted by the Washington timber growers in May to make an aerial map of the infested area. This map showed that the infested trees were not in one tract but were distributed in patches over an area of approximately 20 sq mi. The area to be dusted included 5,000 acres, mostly in three large tracts and a number of small patches.

The service allotted a rugged, varying altitude from sea level to 2,500 ft, and had to go up with reverses. "The forest cover" is very irregular, many extremely high spruces have been seen level against the hemlock. This made dusting both difficult and dangerous as it was necessary to fly close to the trees in order to dust them satisfactorily.

The plane was a converted Ryan Streamliner equipped with a special hopper sprayer driven by an electric motor mounted from a storage battery. Dusting was done by Julius B. Blum, pilot of Northwest Air Service, and by Wesley Gray. Since the affected timber was located near fire, a beach landing field was used. A barge on a float on the beach was loaded from coast of chemical while the plane was in the air, and the mixture was dumped directly into the plane hopper on its return. Fifty-four tons of calcium arsenate, containing a 41 to 42 per cent solution of

oxide of arsenic were used. The plane carried from 500,000 lb of this material on each trip.

Dusting required fourteen days; 137 trips were flown and time in the air totaled 33 hours, 9 min. The average dose on the ground between landing and take-off was 2 min. There were no mishaps.

The results were 66 per cent kill the first day following the dusting; 18 per cent additional kill in the next two or three days, and it was believed that the dusting would account for an additional kill of about 5 per cent, or a total kill of 75 per cent. The average cost per acre was \$2.90.

Fast air survey made in Canada

THE Aerial Survey division of Canadian Airways completed last winter a set of contour maps and oblique photographs and other data for use in engineering a new construction project of the Canadian Power Company at the confluence of the mouth of the Montreal River on Lake Superior. An extremely accurate contour map was produced and the time required was considerably short.

The contract was signed Sept. 30, 1938. The photographic crew—consisting of a pilot, photographer and air engineer and using a Fairchild 71, Fairchild 12-in camera, two objectives, view finder and vertical and speed control—based at the South Bay Marine base on Sept. 23 after a 650-mile flight from Montreal. The photography was

completed after 2 hours 30 min. (Sept. 25, and the plane was recalled Sept. 26, on its return to base). The two roles of film were developed and printed Sept. 27-28 and between Oct. 3 and Oct. 8 a small relief model in plaster was completed from stereoscopic inspection of prints.

Contour work and the permanent maps could not be advanced until the receipt of horizontal and vertical control point data from the coast Dec. 12. This data involved field survey.

The actual contouring took place about the middle of February. The finished mosaic was completed March 6, representing a rapid time since the receipt of the contour data and a half month, two of which were consumed in obtaining control plane. As a result of this experience it appears that the preliminary mosaic may be of great value to the client for ground work in connection with such surveys as location of camps and transportation routes, and for use of field parties in gathering the necessary control data. Such maps may be made within three or four weeks.

Bald Paps used in fly-it-yourself plan

THE Bald Paps have been used in the first of a series of fly-it-yourself plans in the West. Along, Mendocino County Airport, is a successful fly-it-yourself service. One has been flown 206 hours and the other 252 hours at the rate of \$0.10 per mile. On the basis of 26 mph, these two planes have earned a total income of approximately \$2,000 while the operating cost, figured at about \$0.01 per mile, has totaled about \$260.

Very little regarding has been necessary, the company reports, and has associated largely for the service in business experienced by the company this year. An increase of 35 per cent in student enrollment over a year ago, bringing the total enrollment to about 120, is accounted for directly to this service, while a 25 per cent increase in general charter flights has been attributed largely to general interest stimulated by the association. The hourly rate has been reduced from \$4.00, from \$2.00 to \$0.30.

Students are instructed and aided on flights. After a few hours solo they begin their flights by themselves. Most of the persons are non-professional pilots with a private license

Airport Management

Garages for patrons add to port revenue

AN ADDITIONAL source of revenue has been developed by the San Francisco Bay Airports Authority, Cal., through the provision of additional garages for the use of airport employees or airline passengers. Six garages are now in operation. One garage is reserved at all times for the use of trustees and air marshals. There may be some being erected and others will be provided as required.

Transient customers pay a rental fee of 35 cents per night or 55 per week. Field employee discounts pay 35 per week. The garages are provided by night watchmen service. Building and ground on a separate suit located the few hours before and supplies provision of additional service.

Street signal lights control airport traffic

THE New York Municipal Airport—Ford Bennett Field—has lowered a reputation traffic signal from its police department for control of ground traffic on the pier. This installation has the green, amber and red lights on three faces, pointing out toward the main runway. The intersection is on the perimeter of the non-professional business and is taking up the larger line, and is both driven along the runway.

A red light seen from any point along these surfaces holds all traffic in that



New garages at the San Francisco Bay Airports

direction. A green light permits movement along the strip, and take-off or landing. The signal is controlled by the administration building where an attendant is on duty at all hours. This adds another automatic traffic control system to those under study by the Airports Trust and the Airports and Chamber of Commerce.

A service plan for private owners

THE Carter Wright Airport at Valley Stream, L. I., is developing its non-professional business by a new plan of customer service. The management policy is to the satisfaction of the non-professional business and is taking up the larger step to cultivate it. Lower hourly rates is one of the new features. Formerly rates ranged upward from \$40 per month but is a new

possible to house a three-place Travel Air or a Challenger Rains for \$35 a month and the Carter-Jones for \$40.

Not only are these rates lower, but they include some service that has never before been provided as part of the rental. Students will be kept ready, only be taken out on the line and returned on an individual, and when so desired, will be delivered to any airport within the metropolitan area for other duties without an additional charge.

If, for instance, an owner happens to be in Washington County for a week and wishes to fly, he need not notify the airport by telephone the time he wants his plane delivered to the airport at Annapolis. The member is informed there on schedule and is notified for when the owner is through with it.

This service is possible through cooperation of the flying academy. Students bring cross-country training up to the airport pilot as in airport where he has followed a plan. This provides the student with one cross-country trip and one with a passenger for each delivery. The service occurs on a peak-up after the owner is through with his plane, the student takes the customer to the airport and departs solo while the instructor returns with the machine of the private owner.

Another important feature of this plan is the daily students' control. All privately owned aircraft are based in one hangar; the owners will have their own lockers and there will be a hanging room with washing matter and comfortable furniture for their convenience. This will be an change of the chief aviation. Once hangars will be reserved for members of the private events in the need area. At the same several places are being issued to longer No. 5 of the Carter Wright port at Valley Stream.



The recent traffic signal in the Ford Bennett Field and Valley Stream, L. I., is the first of its kind in the country.

What Our Readers Say

"The Patent Racket"

To the Editor:

Your recent editorial "The Patent Racket" appearing in the August issue of *AVIATION* and in an appeal in a direction will do a lot of good. It is my conviction that should have inspired legal aid. The suggestion that external engineering services could pass on the patented merits of ideas as submitted would appear to offer an admirable basis for initiation. Such organizations could no doubt furnish to inventors with means of reputable patent lawyers needed in the aeronautical art. It might go further having reputable attorneys in the respective fields.

The idea of inventors in which editorial refers are as a rule ignorant of the value of patents. They lack knowledge of Patent Office procedure. While this information may be secured from the Office, because of detail it is apt to be found confusing. Information presented to the inventor in other terms giving him an outline of patent office procedure the importance of properly preparing the patents and the moral necessity of lengthy protection in the patent office to insure adequate claims would be more beneficial. A number of terms which most inventors fully comprehensively could be recommended.

While very few patents have any value there are often inventive ideas disclosed to patents that might have helped inventors had the patents been properly obtained. The subject merits a great deal of attention because the problem is so far-reaching; not only in aviation with respect to the so-called "black claim," but I believe you will find professional men at the mercy of unscrupulous lawyers. Perhaps a final or subsequent could be directed to carry on such important work. It doesn't in such matter whether it be in the hands of engineering associates or university extension work or a patentologist's enterprise. Although the former is not in the editorial has the machinery to function.

ROBERT F. HALL
Buckham, N. Y.

To the Editor:

Your editorial on "The Patent racket" in the August issue is a timely one.

The practical complaint of that of taking out patents on obviously superior structures, not only some considerable loss to the defense industry, but

leave the patent procedure into disrepair. Fortunately, the policy necessary can readily be located, thus making the name of the structure appears on the drawings which form part of the patent, and when it does not to suggest it can be obtained from the patent file in Washington.

Another group to blame it seems to me, are the Patent Office examiners who permit such patents to issue. The examiner is supposed to be skilled in the art, and it is his duty to reject an application for a patent on a structure which will not operate to perform the function it is asserted to perform. If instead, a patent is granted, it is likely to be taken as a certificate that the structure is operative, and thus possibly it may assist in fraud.

REV. T. KENNEDY
Knight Brothers
New York City

To the Editor:

I have read with great interest the editorial entitled "The Patent Racket" in the August issue of *AVIATION*.

In the first place, few patent lawyers actually do it. It is usually for the patent lawyer to do so. There are also at patent attorneys, chiefly in Washington, who subvert, and the majority of these clients include inventors in remote parts of the country who do not have direct access to the office of a patent attorney.

In the course of my practice, I very frequently endeavor to convince inventors in many different areas, and particularly in the art of aviation, that their inventions are unpatented. And I have referred such invention to appropriate reputable consulting engineers formerly the inventor is suspicious of such advice, and there has been some cases of a consulting engineer telling in an improper manner to formation obtained from a client of mine. I do not wish to infer that I know of every case of consulting engineers who have improperly advised themselves of information supplied by clients of mine referred to them. Even one such case, however, has an effect on the client which is very difficult to overcome.

When we are suggested to file a patent application, it is usually invention which appears to us as being physically impossible or impossible as far as the client that such is the opening. Very frequently the inventor replies

that he has already tried and the invention is never, and that it works. There is nothing left to do but file the patent application. Otherwise the inventor, after he has been advised, usually enters his departure in common silence.

If an inventor wants to apply for a patent, regardless of the merits or demerits of his alleged invention, the only way to steer him out of his hands is to have a final opinion on his patent application by the Patent Office or an appellate body. After all, this is it should be because the history of invention, as well as the history of all other practical arts, is left of the past, and left unimpaired for years and finally eventually inherent.

In other words, no patent is ever taken as a whole is a valid thing, being based on ideas, and it is a matter of course that the practical solution of many problems has been obtained by men who have used the prior art of unpatented patents. So, after all, the important point is of some value.

We must not forget to the other hand, that the most successful patentee is one who can make his own business activity with the courage to defy that which has been common practice in the past. Research organizations account for most useful improvements on previously known ideas, but there are few examples of any revolutionary inventions coming from such organizations.

Returning to the alleged "Patent Racket," every applicant for a patent concerning a device is examined by the Patent Commission and the Patent Office. The patent attorney merely helps the inventor in completing these rights, and the most thought he may have been opinion as to the terms of any particular invention, his ultimate duty is to make to carry out the instructions of his client. The patent attorney is very rarely able to pick and choose his clients.

The patent process as a whole is constantly endeavoring to overcome the abuses of its black side. It is very difficult to get Congress to pass corrective legislation, but there is a gradual, step-by-step movement going on which may be the last that a great many inventors, however, or later laws that they should expect from a patent attorney and seek the attorney who render the proper service.

W. F. FLEISCH
Fleisch & Skid, Patent Lawyers
Creston, Ohio

[My personal reproducible nature of the process of patent law in the case of the most significant of structures that emerge in it, but there are a certain number of black days who get a stigma upon the white pedestal. It was of course only of these things that we were speaking in the editorial above which Mr. Fleisch writes.—ED.]

Servicing Short Cuts

A PORTABLE BARREL FOR OIL DRAINAGE

REMOVING oil from airplane engine components with little effort and without danger of spilling the drainage on hangar floors has been accomplished in a personal way by the Blackburn servicing unit of the 11th Tactical and Western Airways at Alhambra, Cal., by mounting an ordinary 40-gal oil drum on a small dolly. The drum is fitted with a special strapping device to catch the oil from the tanks, and with a special pipe outlet and valve at the bottom for draining the oil into an underground receiving tank. A single float valve mounted on the side of the barrel minimized in years when it is possible to read the exact amount of oil drained at any one time. It is reported that the use of this equipment effects an average saving of four man-hours a day over the other method of draining into individual containers and carrying away by hand.

WHEEL AND BRAKE SERVICING

FOR minor adjustment of bearings or wheels on the landing gear of the Cessna Kingbirds, used on the Southern



Portable drainage barrel

PRESSURE SYSTEM FOR LUBRICANTS

HANDLING of oil and grease is greatly simplified in the Beeble-Jobs oil-lubed Air-Lites by placing the storage tanks under air pressure so that lubricants may be drawn off at will by merely turning the proper air valve. Two 35-gal oil tanks and a ground tank holding 30 lb. are now controlled. By application of air pressure in the storage tank, oil pumps for ground gear can be filled with great rapidity.

AIRPORT FUEL GAGE

FOR convenience in making accurate and rapid reading of the amount of fuel pumped into the tanks of the Southern airplanes at the Washington Shop of the Langley Labs, a new Wayne gage has been added recently to the fuel dispensing unit (Aviation, June, 1959). The dials are approximately 3 ft. in diameter and is tilted at such an angle that the mechanic filling the tanks high up on the wing of the airplane has a full view of it. The letters are large and can be read much more accurately within the length of the filling hose.



Fuel tank fuel gage

Division of Eastern Air Transport, Inc., the sale of the undercarriage is question is simply raised clear of the ground by means of a hydraulic jack. In cases where the wheel must be removed a soft, well-bored wooden supporting block is pushed under the foot end of the axle and the jack raised until the wheel rests entirely on the support. By reason of its broad base, the support carries the weight without danger of tipping over while other ways are less protected on the plane, and the jack is released for other uses as the shop will the wheel is ready to be removed.

Through the use of such simple and inexpensive specialized equipment, well equipped repair shops are promoting safety by the elimination of hazardous accidents with their attendant risks.

AN EMERGENCY CRASH KIT

AN EXCELLENT emergency crash kit has been developed by the Boeing Airplane Company and is kept available in its hangar on Boeing Field Seattle. The kit weighs about 60 lb., complete and is equipped with handles so that it may be easily carried by one or two men. The tools include wrench, screwdriver, nut driver, heavy wire and trim for outer adjustable brackets, saw, and an edge bender. Mechanics are furnished to get them for engine work but they prefer to be available when required in an emergency.

TEXACO AVIATION GASOLINE FOR AMERICAN AIRWAYS, INC.



Today one can travel by American Airways planes comfortably from New York to Los Angeles in 29 flying hours.



THE Southern Lines of the American Airways use Texaco Aviation Gasoline exclusively at terminals from Atlanta to El Paso.

The American Airways operates a nationwide system of air transportation for passengers, express and the United States mails. Its planes are of the most modern type, superbly designed for the safety and comfort of passengers. Pilots are in constant touch by radio en route with 200 government weather

stations and 65 special American Airways stations. Unusual airport inspection facilities are provided to insure safe arrival at destinations and unbroken schedules.

In line with this high type of service, twenty-four of the American Airways' Southern Air Terminals are stocked exclusively with Texaco Aviation Gasoline. The operating staffs at these terminals are enthusiastic boosters of Texaco Aviation Gasoline.

This super-power gasoline is available at the Country's principal airports and landing fields from coast to coast.

THE TEXACO COMPANY, 1200 BROADWAY, NEW YORK CITY



TEXACO AVIATION GASOLINE
TEXACO AERODIESEL FUEL
TEXACO AIRPLANE OILS
TEXACO MARFAK GREASES
TEXACO ASPHALT PRODUCTS
FOR RUNWAYS, HANGAR FLOORS AND
AIRFIELDS, AND BODY LATHING.



DON'T FLY ANOTHER WINTER WITHOUT AIRWHEELS

MOST of the year's worst weather comes in the next five months.

You may be able to pick your ground conditions at the start of a flight—but nobody knows what you'll find when you come back to earth again.

Play safe—with the only wheel and tire equipment soft enough to land safely on mud or snow. Equip yourself with Goodyear Airwheels.

The cost of Goodyear Airwheels and the new Airwheel rubber bearing brakes is almost exactly the same as you'd pay for wheel and tire equipment far inferior in safety.

They're worth having, just for their protection in emergency landings—but they mean easier flying, and they save your ship in every-day service.

Ground loops are almost impossible—you can get far better take-offs and landings with Airwheel brakes—cross-wind or down-wind landings are easy—you save your ship from the vibrations of landing over rough fields—when you have these big, soft, rolling rubber pillows between you and the ground.

Only Goodyear can give you Airwheel safety. For full proof of what Airwheels can do—write to Goodyear, Akron, Ohio, or Los Angeles, California.

**WHEN YOU BUY A NEW SHIP
SPECIFY GOODYEAR AIRWHEELS**

GOOD YEAR

EVERYTHING IN RUBBER FOR THE AIRPLANE





The Most Sensational Performance in the History of the Air Races...

Richfield Wins the 3 Major Events with a New World's Record in Every One!

THE Thompson Trophy Race won by Lowell Bayler; the Aermotor Trophy Race won by Maud Tait; the Invitational Free-for-all won by Robert Hall. All won with Combat Richfield Aviation Gasoline. Bayler's speed of 256.239 miles per hour exceeds all existing records for a land plane over a closed course, and topped the previous world's record made only the day before by Robert Hall in the Invitational Race by 14 m. p. h. Miss Tait set a new world's record for women of 187.574 m. p. h.

That... and numerous other closed course events credited to Richfield products... mark up Richfield's brilliant performance at the National Air Races this year. It indicates the wide margin of victory over all other gasolines; no other motor fuel even remotely approached this record during the races.

Since the inception of the National Air Races... Richfield has been the outstanding winner among gasolines. It must be a superior product to enjoy the preference of pilots who are capable of new world's records... amazing victories in these world-famous tests of speed and stamina.

The Corbett Richfield for your own plane... get this superior performance... at important airports both east and west of the Mississippi.

COMBAT
Richfield's Aviation
GASOLINE

RICHFIELD OIL COMPANY • LOS ANGELES • NEW YORK CITY

AGAIN... AMERICA'S SPEED CLASSIC WON WITH WASP JUNIOR



Pratt & Whitney Engines Gain a Lap in Leadership at 1931 National Air Races

Clasifying a series of performances that brought new honors to Pratt & Whitney at Cleveland, Wasp engines swept home with both honor and money—top in the Thompson Trophy Race. Lowell Bayler in his Wasp Junior powered Gipsy. He averaged 256.239 m. p. h.—a record consecutive victory for the Wasp Junior in America's greatest speed classic. Followed him closely came J. H. Wootton in a Wasp Junior—also equipped with a Wasp Junior.

Earlier in the meet Wasp engines took the first six places in the Bendix Cross Country Trophy Race from California led by James Doolittle in his land "Solitaire." He averaged over 253 m. p. h. to Cleveland, then continued on to New York and a new World-East, Coast-to-Coast record of 11

hrs., 16 min., 10 sec. Likewise, in the Women's Free-For-All Classic, Wasp power captured first and third places. Maud Tait set a Wasp Gipsy Race Senior Sport led Miss Haskin flying a Wasp Junior land. These are but a few of the outstanding events of this year's races in which Wasp scored significant triumphs.

Speed—never a sufficient goal in itself—is attained in Pratt & Whitney engines by coordinated substantiation in design and construction that likewise increases their reliability and power. The high performance noted above are dramatic verification of the qualities that have made Pratt & Whitney engines the choice of racing, civil, military pilots and more than 90% of all important transport lines on this continent.



Lowell P. Bayler and the Wasp Junior that helped him capture the Thompson Trophy

and
PRATT & WHITNEY AIRCRAFT CO.
DAYTON, OHIO • U. S. A.

Manufactured under license by Canadian Pratt & Whitney Limited, Montreal, Canada; by Pratt & Whitney Aircraft Co., Ltd., Longford, Ontario; by Canadian Empire, by Eastern States Aircraft, Memphis, and by Japan by Nakajima Aircraft Works, Tokyo.

Wasp & Hornet Engines

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A TRIBUTE TO THE NATIONAL AIR RACES

Planes racing at terrific speed—starts so daring that experienced flyers hold their breath—meets of aircraft representing many nations daily clouding the skies—and not a fumble during the whole meet—a splendid tribute to Aviation and the National Air Races Management. Parachutes played an important part in making this record as four crashes were saved from being tragedies when the pilots jumped and saved their lives by their parachutes.

A high light of the meet was the new altitude record set by Switlik representative, Bert White, in a Switlik Safety Chute—again proving that from high or low altitude, the famous Switlik Safety Chutes give the perfect performance.

Switlik Chutes are sold on an EASY PAYMENT PLAN for your convenience. Write today.

*Used by the Navy, Dept. of Commerce
Air Mail, and Many Famous Flyers*

SWITLIK SAFETY CHUTES

SWITLIK PARACHUTE & EQUIPMENT CO.
TRENTON, NEW JERSEY

WESTERN MANGER — BIRD WHITE — 1013 AIRWAY — GRAND CENTRAL AIR TERMINAL — GLENDALE, CALIF.



Look to BENDIX for Airplane Wheels and Brakes

Bendix engineering developed and perfected the wheel-and-brake for aircraft—which has contributed so vitally to efficiency in ground maneuvers; and to safety.

Bendix now offers the new roller-bearing wheel and brake for still further increased efficiency.

Bendix also offers the Bendix low-pressure wheel, equipped with roller bearings.

All Bendix' broad experience in brakes—automobile and aviation—is built into these new units.

Built to U. S. Army Air Corps, U. S. Navy, S. A. E., and Tire and Rim Association Standards.

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FOR SAFETY

FULLY PROTECTED BY PATENTS AND APPLICATIONS IN U. S. AND ABROAD

See Bendix Sales Office



Do a little "GROUND FLYING"



Check up on your oil as carefully as you
do your ship and motor

TALK to the pilot who uses Gulfpride Oil and he will tell you that his experience squares with the findings of leading aviation engineers.

Gulfpride Oil shows measurably low viscosity-temperature curves... they hold their body under high operating temperatures, yet they never become rubbery stiff and sticky at low starting temperatures.

Gulfpride Oil's fuel oil is the most work carbon.

Gulfpride Oil gives you more power from your motor.

Technicians can tell you that Gulfpride Oil is valued by the Aluminum Chloride process, which employs motors costing \$100 per ton instead of the \$60 per ton for commonly used.

That explains why pilots who demand the best in lubrication use Gulfpride Oil.

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Over 35,000 copies of the Gulf Aviation Index have been distributed. Nine editions, new model, larger models and historical maps, charts, etc. up to date. Send five or more to demand copies. Limited quantities, all those interested with the greatest interest will be given priority. Write to: Gulfpride Oil, Gulf Refining Company, General Sales Office, Pittsburgh, Pa., U. S. A.

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Gulfpride Oil is 100% grade four motor oil refined by the new Aluminum Chloride process. Tests show in Gulfpride Oil carbon deposits are reduced to a minimum for maximum engine life and low fuel costs.

Gulfpride Oil holds its body as viscosity-temperature curves. It is highly resistant to oxidation and emulsification under Gulfpride Oil is made in five viscosity grades: 70-110-120-150 and 200 at 100°F. It meets the highest requirements for all types of aircraft engines, and is the only oil that meets the requirements of the Federal Aviation Commission.



**Gulfpride
Oil**

• GULF REFINING COMPANY •



TAKE for instance, upside down flying. Naturally, the same force that causes the pilot to be supported by his belt, causes the fuel to go to the top of the float chamber of the carburetor.

To prevent excess fuel going to the engine, Stromberg uses a check valve, incorporated in the needle valve seat. Stromberg Aircraft Carburetors function perfectly in the inverted position, the engine continues to run at full throttle. Fuel discharge needles, located in line centrally with the center of the float, keep the fuel flow from being disturbed.

When power is demanded from an engine



equipped with a Stromberg, it is obtained regardless of the maneuver or position of the plane.

Assurance of power all ways is just another Stromberg contribution to the advancement of flying. Stromberg's 22 years experience and research in carburetors is years in comparison.

Over 50% of the aircraft engines being built in the United States today are Stromberg equipped.

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"Happy Landings!" Of course! But insure "Happy Landings" by fueling your plane with Socony Aviation Gasoline and Socony De-waxed Motor Oil. These products are air-tailored by our own aviation experts—men who test them continually in Socony's own Test Plane under the actual conditions an average pilot encounters. Use this combination every time—Socony Aviation Gasoline and Socony De-waxed Motor Oil—and you'll guarantee yourself "Happy Landings."

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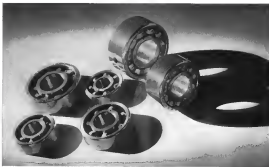
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Nickel Cast Iron cylinder for Wright Gypsy 4-cylinder engine sold by J. CRONIN & SONS, Boston, N. Y.

Below: Wright Gypsy vertical 4-cylinder in-line 200 H. P. airplane engine sold by WRIGHT AERO-NAVIGATOR CORPORATION, Patuxent, New Jersey



WRIGHT KNOWS NICKEL CAST IRON CYLINDERS MEAN BETTER ENGINE PERFORMANCE

When a longer wearing cylinder material was sought for the Gypsy—the latest Wright vertical 4-cylinder in-line engine designed for smaller craft—Nickel Cast Iron was the most common choice of Wright engineers.

Nickel Cast Iron, called "the longest wearing cylinder material known today," has unquestionably contributed to the successful performance of the Wright Gypsy. Long hours of flying at high speeds demand a constant strength and dense throughout all sections...with high uniform hardness assuring exceptional wear-resistance in cylinder bores. These properties Nickel Cast Iron provides, not only in Wright engines, but in the cylinders of approximately fifteen other leading airplane engines as well.

THE INTERNATIONAL NICKEL COMPANY, INC., 47 WALL STREET, NEW YORK, N. Y.
Blowers, engines and valves of Nickel—the producers of Nickel Steel



OUR CUTTING SPECIALISTS WILL GLADLY DESIGN YOUR PROGRAMS WITH THE

A BETTER MATERIAL FOR MODERN MACHINES

Stearman steps ahead



Obvious! you want a good ship—a ship that will go places—and quite as obvious! you would expect service facilities for that ship whenever you go... on-the-spot service at convenient ports—nations-wide... where skilled hands... men familiar with your ship are at your service. This is what. This is flying's next step forward... Stearman has already taken it. Wise buyers are alert... You'll be seeing more Stearmans in the air.



STEARMAN AIRCRAFT COMPANY,
WICHITA, KANSAS, Division of United
Aircraft and Transport Corp.

For every flying reason
STEARMAN



- The Stearman ship itself
- The Stearman organization
- Backed by the United Group
- The new biplane, built at million-mile service facilities



EN ROUTE

...AND IN TOUCH
WITH THE WORLD

...THE PLANE IS A POSTAL TELEGRAPH OFFICE

Like all other creatures, the machines have enough to send us news. A message—an appointment—whatever the message is, they expect the same facilities for sending it from a plane as they do from a room or a ship. And why shouldn't they have the service, especially when it means no expense to the transport company?

If you need to go back your plane in a world-wide system of communication facilities it is possible. Postal Telegraph Radio in each plane will connect your airplane to the messages or scheduled stops.

The passenger wants the message to his home. The attention also is at the most scheduled stop. Postal Telegraph Radio in the air. Simple, isn't it? And it's a convenience, even service the passenger appreciates—a service that makes us travel more agreeable.

Postal Telegraph Radio is a part of the world's largest communication system in flight. It is the only system in the world that offers a world-wide service of communication under a single management. Through the great international system of which Postal Telegraph is a part, it reaches Europe, Asia, the Orient, even Commercial Cables, Central America, South America and the West Indies over All America Cables, and ships it to the Mailing Radio.

Postal Telegraph Radio is a part of the world's largest communication system in flight. It is the only system in the world that offers a world-wide service of communication under a single management. Through the great international system of which Postal Telegraph is a part, it reaches Europe, Asia, the Orient, even Commercial Cables, Central America, South America and the West Indies over All America Cables, and ships it to the Mailing Radio.



THE INTERNATIONAL SYSTEM

Postal Telegraph

Commercial
Cables



All America
Cables

Mailing Radio

WHEN YOU GO OUT, COSTS GO DOWN SO OUT YOU GO, MR. WATER-THIN!



Mr. WATER-THIN is a loafer. A waster. A spender of your money. For he's the quart or more of that, wasted, ordinary refining process in every gallon of motor oil. Non-lubricating stuff that vaporizes quickly under engine heat.

But, there's more of this waste in Quaker State Aero Oil. Quaker State refining—the most modern in the industry—removes it. Removes it and replaces it with rich, full-bodied lubricant. Quaker State gives you

four quarts of good oil to the gallon—not three quarts and one of waste. So you really get an extra quart. You get an oil so good that the demand for it has made Quaker State the world's largest selling Pennsylvania Oil!

Every drop of Quaker State Aero Oil is made entirely from 100% pure Pennsylvania Grade Crude Oil. Quaker State is so free from impurities that it doesn't require any treatment in refining. That's impor-

tant! For acids tend to destroy some of an oil's efficacy.

Test out Quaker State Aero Oil in your planes. You will find that it stays good lubrication long after ordinary oil would be reduced to useless waste. You will find that it saves money in oil costs and in maintenance costs. You will discover right away that Quaker State's extra quart of lubrication makes it the finest oil, the most economical oil that ever went afloat!

QUAKER STATE MOTOR OIL



THERE'S AN
EXTRA QUART
OF LUBRICATION
IN EVERY GALLON

They put *Curtiss Conquerors* in the DO-X —but let Capt. Fr. Christiansen tell the story



... and since this letter was written by Capt. Christiansen the DO-X on its trade missionary trip has flown from Brazil, touched at Cuba and along the Eastern Coast of the United States and successfully landed at New York City. The world-traveling DO-X adds fame to the Dornier staff that designed and built it. Every one of the thousands of miles it has flown adds to the Curtiss-Wright reputation for building Aviation's most reliable power. When the extreme world trip was planned through the

chill of Alpine heights and the heat of Equatorial rain, with true German thoroughness the most reliable power was sought. From the world market, 12 Curtiss-Wright "Conquerors," of 600 h.p. each, were chosen. Colonel Lindbergh, as a result of his experience with the reliability of Wright Engines, selected a Wright for his flight to Japan. It is this same reliability that has crisscrossed *Wright-Powered Planes* in over the National Reliability Tour six consecutive years.



WRIGHT

AERONAUTICAL CORPORATION
FATERTON NEW JERSEY
A DIVISION OF CRITCHFIELD-WRIGHT CORPORATION

CORSAIRS

*that wear the
GLOBE AND ANCHOR
of the U. S. Marines*



Corsairs in service with the Marine Corps are usually in motor service. It may be in Haiti, whose rugged, mountainous interior affords few landing fields that are even possible. It may be one of the inaccessible jungles of Central America. It may be in China in sections where plane fathers would write "Fata" to an aerial mission.

Corsairs risk no odds of the men who make these assignments, or the men who do the flying. The stamina to stand rough landings and the performance to get into and out of small fields are trademarks with this plane. So, too, are its speed—its climb and its excellent landing qualities.

These distinctly Vought characteristics have carried Corsairs through years of strenuous service with the Marine Corps. They have made the Corsair a standard observation plane with the Navy. And they make it a ideal ship for fast extensive transport and private flying. Chance Vought Corporation, Division of United Aircraft & Transport Corporation, East Hartford, Connecticut. Export representative: United Aircraft Export, Inc., 230 Park Avenue, New York, N. Y.



**CHANCE VUGHT
CORPORATION**



Fig. 11 of a series of photographs on "Rope-Making" by Roebling. Shot on Reeling # 100 High.

World's largest rope-making machine

"World's Largest" does not begin to reveal the capability or tremendous capacity of this giant—Roebling machine, that takes huge strands of steel wire and "lays" them into finished rope. In rope diameter, the capacity of this machine is practically unlimited. It can make ropes of a diameter far beyond commercial use. In fact, about its only limitation is a capacity to now cut a maximum of 50 tons in weight of rope in a single length without rethreading. And such a load exceeds the normal capacity of cranes. This "laying machine", one of many of various types used by Roebling, is an excellent ex-

ample of Roebling's unsurpassed rope making facilities. With such a machine, rope of the largest diameters can be handled without the slightest danger of over-stressing. The result is—higher rope efficiency, higher all-around quality.

Incidentally, all the large suspension ropes for the famous George Washington Memorial Bridge, spanning the Hudson River, were made in the machine shown.

JOHN A. ROEBLING SONS CO., TRINTON, N. J.
Roebling's Principal Cords: *Extra Size*—New York
WIRE—HOLDING WIRE—TIE WIRE—CABLES—STEEL CABLES—STEEL WIRE
WIRE AND CABLES—WIRE CABLES—STEEL WIRE CABLES

ROEBLING WIRE AIRCRAFT PRODUCTS

What is "OCTANE NUMBER"?

What does it mean to pilots
and to men who buy gasoline?

Engine reliability is essential to safety in the air. Engineers everywhere agree that no single factor has so vital an effect on engine performance as the "knock-rating" (or anti-knock value) of the fuel used. Knocking or detonation will ruin the finest of engines.

All crude petroleum and all gasoline made from them consist chiefly of hydrocarbons, chemical combinations of hydrogen and carbon. One of these, iso-octane, has an exceptionally good knock-rating; another, normal heptane, exhibits the opposite characteristic. By mixing the two in various proportions, a fuel of any desired intermediate knock-rating may be obtained.

The octane number is the generally accepted method of expressing the knock-rating of a gasoline. It simply means the percentage of octane in the octane-heptane mixture which has the same knock-rating as the gasoline. As

iso-octane is the knock-suppressor in the mixture, the higher the octane number of a gasoline the better its knock-rating, or anti-knock value.

To the pilot or the purchasing agent buying aviation gasoline, the octane number furnishes a definite measure of the most essential characteristic. A gasoline having a minimum octane number of 73* will allow full throttle operation without destructive detonation in the very great majority of present-day, air-cooled aviation engines. Stanavo distributors or the manufacturer of your engine will gladly advise you regarding the fuel requirements of your engine.

*This value is based on tests using the series 30 Ethyl Gasoline Corporation single cylinder test engine at 600 r.p.m., 200° F. jacket temperature and 15° spark advance.



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Organized and Maintained by

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222 Bank Bldg., San Francisco

Standard Oil Company (Indiana)
200 W. Michigan Ave., Chicago

Standard Oil Company of New Jersey
30 Broadway, New York City

The PILGRIM 100-A

Transport Airplane

Developed to suit today's
transportation needs — not
designer's prejudice



In every kind of transportation!

Passengers demand COMFORT

They choose their travel service on that basis. A study of comments from more than 100,000 air travelers has helped build this "passenger's" airplane. That's why it has ample leg room • wide easy seats • individual ventilation control • individual heating control • fullview windows • quietness • ample head room • toilet facilities • no bother from baggage and mail loading • maximum speed.

The ability to provide swift, comfortable, careful travel and the features necessary for maximum cost and an operating profit, are combined in the design of the

Operators demand PROFIT

They choose equipment on that basis. Study of more than 10 million miles of country-wide airline operations has helped build this "operator's" airplane. That's why it carries 9 passengers • their baggage • 115 pounds of mail • (space for 135 pounds) • two-way radio telephone and radio beacon equipment • night flying equipment • all power navigation aids • fuel for over 400 miles • at a speed of 2 miles per minute.

PILGRIM 100-A Transport Airplane

AMERICAN AIRPLANE & ENGINE CORPORATION

Manufacturing Division of The Avco Corporation

Farmingdale, Long Island, New York

Manufacturers of PILGRIM Airplane and RANGER Engine



First class of a lot of 10 PILGRIM 100-A Airplane, ready to leave
Farmingdale for Albany in American Airlines

.... KENDALL RIDES

... In the Transcontinental
the 1931 National Air Races,
by Pilots Winning Over 86%



D. C. Warren, first prize
winner of the 1931 National
Air Races

Warren's flight the first
of Cleveland's first race
and endurance machine

Never before in Aviation
history has any product
achieved such overwhelming
approval as the first
and open choice of Ken-
dall Oil by pilots, both
men and women, who

participated in the National Air Races of 1931. Never before
has the outstanding worth of a lubricating oil been
so thoroughly demonstrated as by the splendid victories in
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Transcontinental Handicap Derbies

For the seven part of speed and endurance dominated in the Cal-
ifornia to Cleveland Derby for men and for women pilots, Kendall
was the lubricant used in 10 of the 41 competing planes. The
results were all Kendall.

Women's Derby	First	Second	Third	Engine
First	Phyllis Quirk	Phyllis Quirk	Phyllis Quirk	Wright
Second	May Haskin	May Haskin	May Haskin	Wright
Third	May Haskin	May Haskin	May Haskin	Wright
Men's Derby	First	Second	Third	Engine
First	D. C. Warren	D. C. Warren	D. C. Warren	Wright
Second	Lee Brown	Lee Brown	Lee Brown	Wright
Third	Edwin Goetz	Edwin Goetz	Edwin Goetz	Wright



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Transcontinental Speed Classics

In the Thompson Trophy Race for men, and the Cleveland
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the winners from start to finish.

Thompson Trophy	First	Place	Engine
First	Loeffl Bayle	Gay Rex	Wright
Second	J. B. Widdell	Walter Williams	Wright
Third	Bob Jackson	Lord "Solomon"	Wright
Cleveland, Pennsylvania Aerial Trophy			
First	Maude Tait	Gay Rex "Y"	Wright
Second	May Haskin	Lord	Wright
Third	Phyllis Quirk	Gaines	Kinner

Kendall—All the Way

The early results show that in all the events, including the
Derbies, there were 121 winners and of these 278 planes were lubri-
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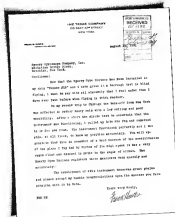
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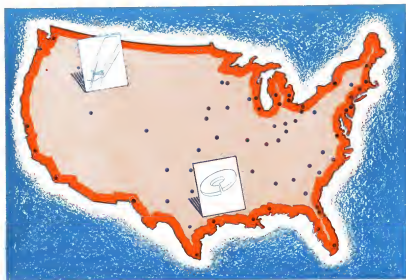
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